

USING MACHINE LEARNING FOR FORMULATING NEW WALL  
FUNCTIONS FOR DETACHED EDDY SIMULATION: PRESENTED AT  
ETMM14 [2]

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  - Machine learning methods such as Support Vector Machines (**SVM**) and **neural networks** are used for solving this type of problems.
  - Through as **much data** as possible at ML?
- In my case, input and output are **numerical** values. **Regression** methods (SVR or NN) should be used [3]; I use **support vector regression** (SVR) methods in Python.



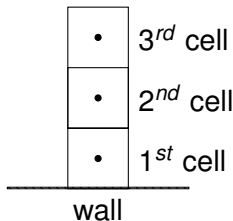
## TRAINING: I NEED A TARGET DATABASE

$$\frac{\partial \bar{v}_i}{\partial x_i} = 0$$
$$\frac{\partial \bar{v}_i}{\partial t} + \frac{\partial}{\partial x_j} (\bar{v}_i \bar{v}_j) = -\frac{\partial \bar{p}}{\partial x_i} + \frac{\partial}{\partial x_j} \left[ (\nu + \nu_{sgs}) \frac{\partial \bar{v}_i}{\partial x_j} \right]$$

- Fully-developed Channel flow
- IDDES.  $96 \times 96 \times 96$ , Reynolds number is 5 200
- Database: average in  $x$  and  $z$

$$\bar{U}_{1st}(x, z) = \frac{1}{\Delta X \Delta Z} \int_{x,z}^{x+\Delta X, z+\Delta Z} \bar{u} dx dz$$
$$\bar{u}_\tau(x, z) = \frac{1}{\Delta X \Delta Z} \int_{x,z}^{x+\Delta X, z+\Delta Z} u_\tau dx dz$$

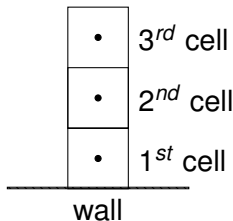
- LES with wall functions:  
the object is to develop a model for the wall shear stress,  $\tau_w = \rho u_\tau^2$



1 <sup>st</sup> cell	$\langle \Delta y^+ \rangle$
Location 1	12
Location 2	31
Location 3	49
Location 4	66
Location 5	76
Location 6	88
Location 7	135
Location 8	155
Location 9	207

300 independent instantaneous samples of  $\bar{U}$  stored at all  $3 \times 9$  cells

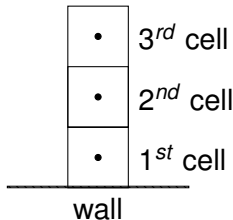
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- Input data:  $U_P, y_P,$   
 $\partial \bar{U} / \partial y, \partial^2 \bar{U} / \partial^2 y$



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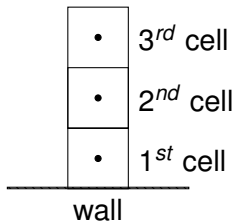
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- output data:  $u_\tau$



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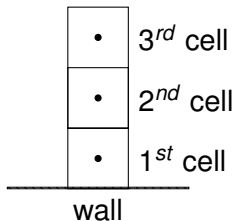
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- Non-dimensional:  $\frac{u_\tau}{\langle u_\tau \rangle} =$   
 $f(Re, y^+, T \partial \bar{U} / \partial y, \partial^2 \bar{U} / \partial y^2 / (\bar{U} T^2))$



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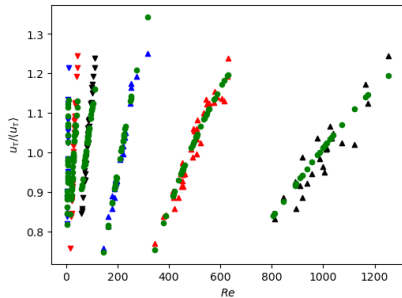
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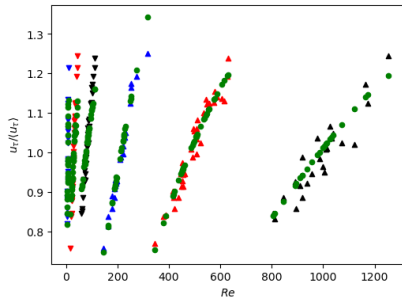
# PREDICTED OUTPUT USING ML: 1ST ATTEMPT



- Output on y axis

(A) ▲: IDDES, Location 1; ▲: IDDES, Location 2; ▲: IDDES, Location 3; ▼: IDDES, Location 4; ▼: IDDES, Location 5; ▼: IDDES, Location 6. ○:  $svr$ .

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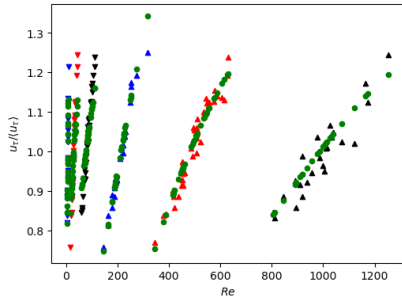


- Output on y axis
- Input on x axis

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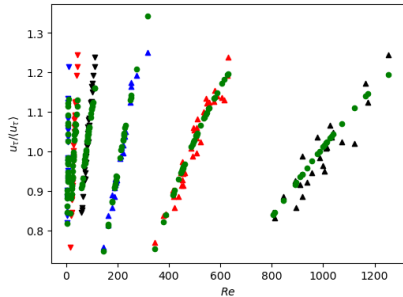
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- Output on y axis
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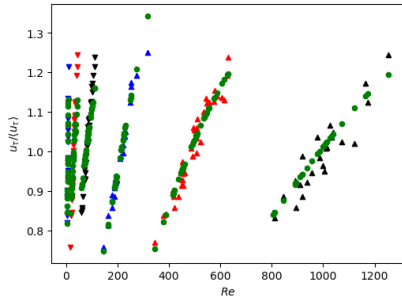
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- Output on y axis
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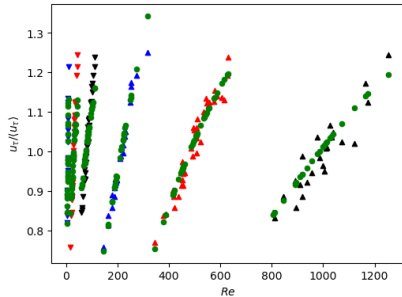
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- Output on y axis
- Input on x axis
- Location 1 – 6 of data
- Difficult to interpolate
- Remedy: I included  $\langle y^+ \rangle$  as input parameter = Location
- $\frac{u_\tau}{\langle u_\tau \rangle} = f(Re, \langle y^+ \rangle)$

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- Traditional wall laws:  $\frac{U}{u_\tau} = f\left(\frac{u_\tau y}{\nu}\right)$

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$C_\mu^{-1/2} u_\tau^2$  : fix  $k$

$\frac{u_\tau^3}{\kappa y}$  : fix  $\varepsilon$



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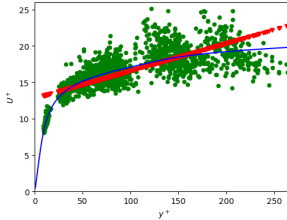
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— :  $\langle \bar{u} \rangle$ , IDDES; ▼: svrLINEAR; ●: IDDES, test data. 9% normalized error.

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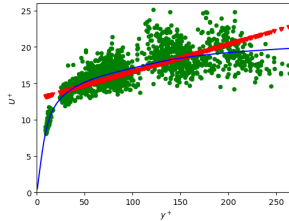
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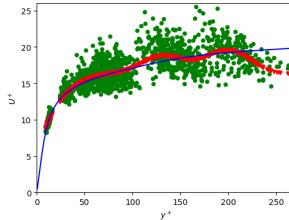
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# STANDARD WALL FUNCTIONS

- The machine-learning wall functions will be compared to wall functions based on Reichardt's law

$$\frac{\bar{u}_P}{u_\tau} \equiv U^+ = \frac{1}{\kappa} \ln(1 - 0.4y^+) + 7.8 \left[ 1 - \exp(-y^+/11) - (y^+/11) \exp(-y^+/3) \right]$$

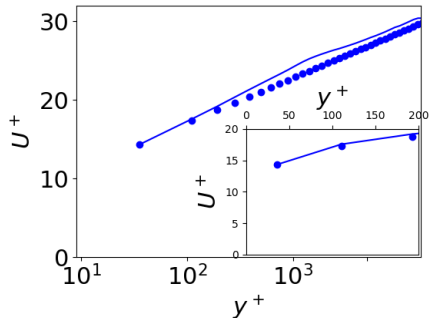
- The friction velocity is then obtained by solving the implicit equation

$$u_\tau - \bar{u}_P \left( \ln(1 - 0.4y^+)/\kappa + 7.8 \left[ 1 - \exp(-y^+/11) - (y^+/11) \exp(-y^+/3) \right] \right)^{-1} = 0$$

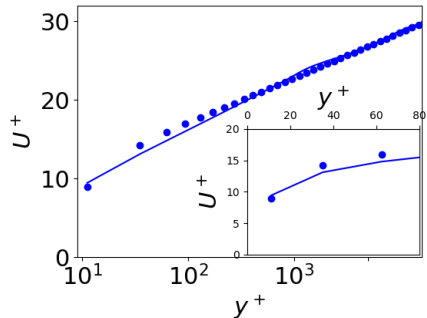
using the Newton-Raphson method `scipy.optimize.newton` in Python.

- $\bar{u}_P$  denotes the wall-parallel velocity in the **first**, **second** or **third** wall-adjacent cell.

# RESULTS, CHANNEL FLOW, ML, $Re_\tau = 16\,000$



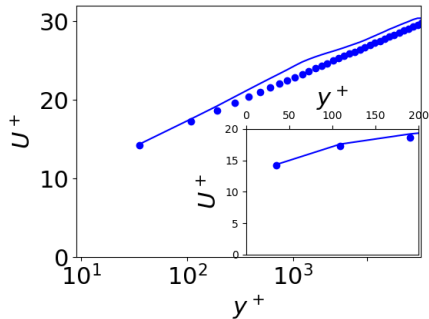
(A)  $N_y = 66$ , stretching 11%.



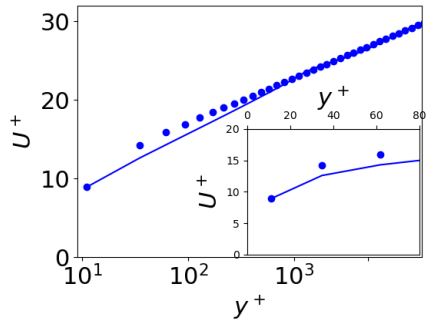
(B)  $N_y = 68$ , stretching 14.7%.

FIGURE: Channel flow. *svr*.  $Re_\tau = 16\,000$ . Velocity. •: Reichardt's law.

# REICHARDT'S WALL FUNCTION, $Re_\tau = 16\,000$



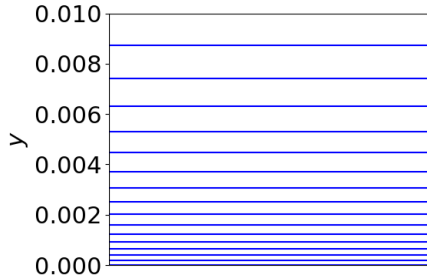
(A)  $N_y = 66$ , stretching 11%.



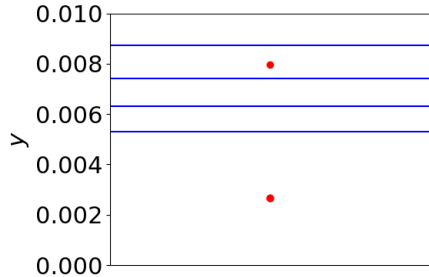
(B)  $N_y = 68$ , stretching 14.7%.

FIGURE: Channel flow. Reichardt's wall function.  $Re_\tau = 16\,000$ . Velocity. •: Reichardt's law.

# NEW GRID STRATEGY



(A) Low-Re number IDDES grid.

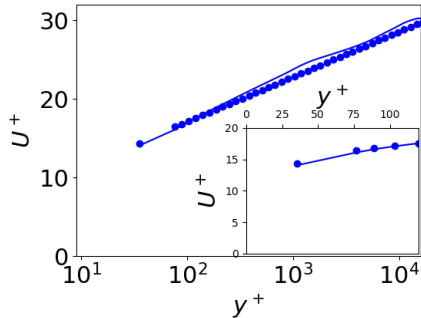


(B) Wall function grid. New grid strategy.

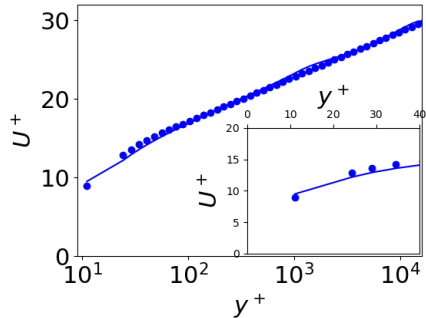
FIGURE: Different grids. — : grid lines.

- This strategy was used in [1] for channel flow and impinging jets (RANS)

# CHANNEL FLOW, ML, $Re_\tau = 16\,000$ , NEW GRID



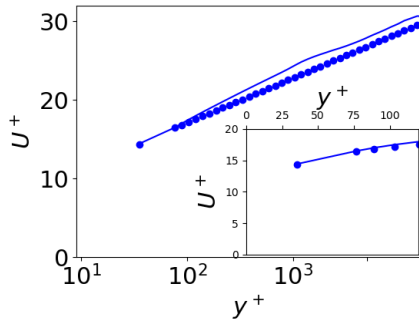
(A)  $N_y = 78$ .



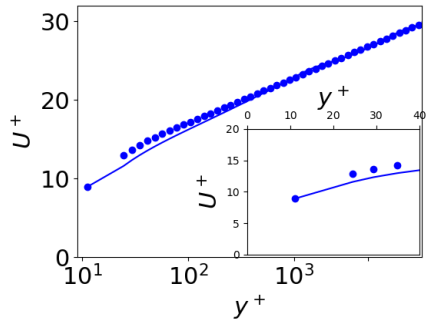
(B)  $N_y = 92$ .

FIGURE: Channel flow.  $Re_\tau = 16\,000$ . Velocity. svr. •: Reichardt's law.

# REICHARDT'S WALL FUNCTION, $Re_\tau = 16\,000$



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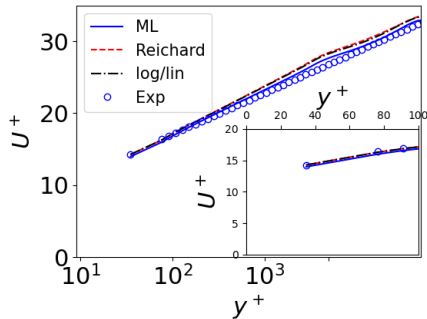


(B)  $N_y = 92$ .

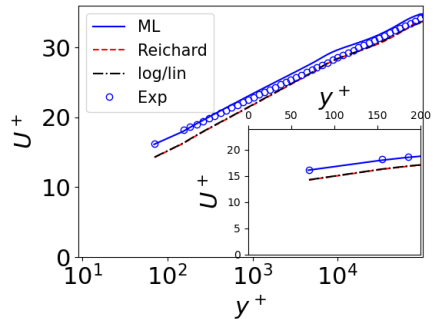
FIGURE: Channel flow.  $Re_\tau = 16\,000$ . Velocity. Reichardt's wall function. •: Reichardt's law.



# CHANNEL FLOW, ML, $Re_\tau = 50\,000$ AND $Re_\tau = 100\,000$



(A)  $N_y = 92$ .  $y_1^+ = 35$ .  $\frac{(\Delta y)_1}{(\Delta y)_2} \simeq 5$   $Re_\tau = 50\,000$ .



(B)  $N_y = 92$ .  $y_1^+ = 70$ .  $\frac{(\Delta y)_1}{(\Delta y)_2} \simeq 5$   $Re_\tau = 100\,000$ .

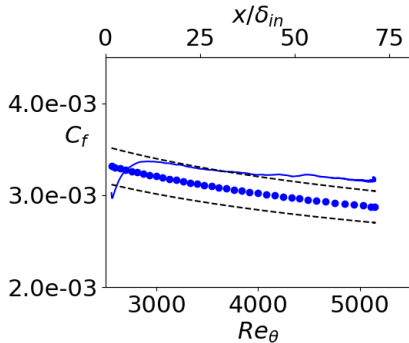
FIGURE: Channel flow. Velocity.

# DEVELOPING BOUNDARY LAYER FLOW

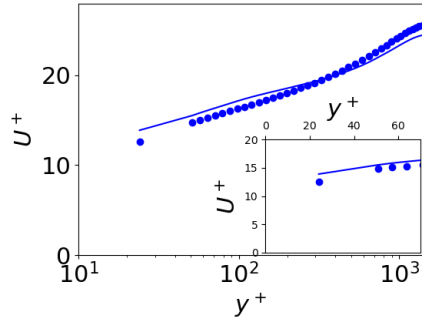
- $Re_\theta = U_{free}\theta/\nu = 2\,550$  at the inlet.
- Domain  $(96 \times 7 \times 5)\delta_{in}$ .
- Grid  $(550 \times 82 \times 64)$ .

# DEVELOPING BOUNDARY LAYER FLOW

- $u_\tau$  computed using 3<sup>rd</sup> cell



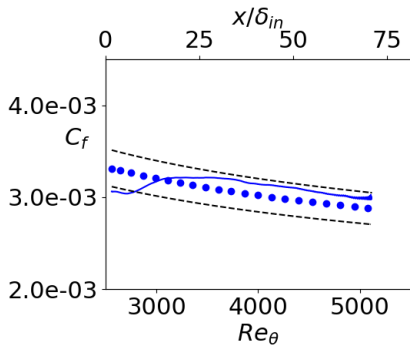
(A) Skin friction.



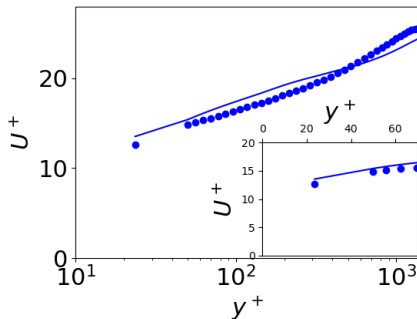
(B) Velocity at  $Re_\theta = 4000$ . Markers: DNS [6]

FIGURE: Boundary layer flow.  $Re_\theta = 2500$  at inlet.  $svr$ .  $N_y = 82$

# DEVELOPING BOUNDARY LAYER FLOW, $2\Delta x, 2\Delta z$



(A) Skin friction.

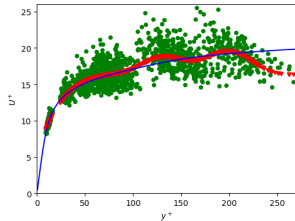


(B) Velocity at  $Re_\theta = 4\,000$ . Markers: DNS [6]

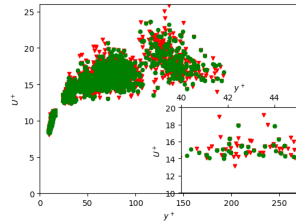
FIGURE: Boundary layer flow. *svr*.  $N_y = 82$ ,  $N_k = 32$ ,  $\Delta x_{in} = 2\Delta x_{in,base}$

# ATTEMPT 3

- Instantaneous data are used for training `svr`
- `svr` finds the time-averaged regression line (shown by ▼ in Fig. A)
- If I want *instantaneous*  $u_\tau$ , I could find nearest neighbour (shown by ● in Fig. B)



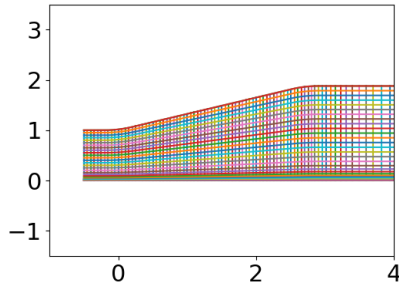
(A) — :  $\langle \bar{u} \rangle$ , IDDES; ▼: `svr`; ●: IDDES, target data. 9% normalized error.



(B) Nearest neighbor using Python's `scipy.spatial.KDTree` ▼: `KDTree`; ●: IDDES, target data; 0.7% normalized error.

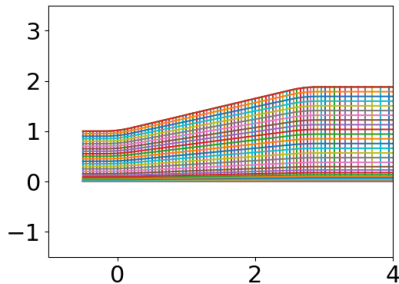
## NEXT STEP: DIFFUSER, PRESSURE GRADIENT

- Inlet: precursor wall-resolved LES of flow in a half-channel at  $Re_\tau = 2\,000$  ( $Re_b = 50\,000$ ).  $600 \times 150 \times 300$ ,  $0.3 < \Delta y^+ < 22$ ,  $\Delta z^+ = 11$ ,  $\Delta x^+ = 22$



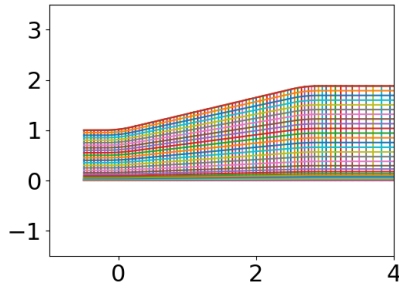
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- Diffuser: same mesh at inlet; for  $x \geq 2.9$ , 1.005% stretching.



## NEXT STEP: DIFFUSER, PRESSURE GRADIENT

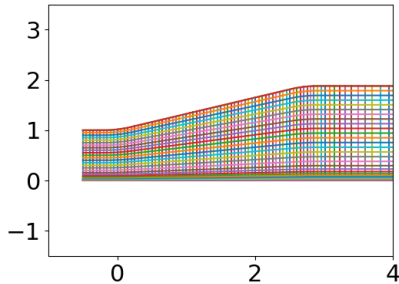
- Inlet: precursor wall-resolved LES of flow in a half-channel at  $Re_\tau = 2\,000$  ( $Re_b = 50\,000$ ).  $600 \times 150 \times 300$ ,  $0.3 < \Delta y^+ < 22$ ,  $\Delta z^+ = 11$ ,  $\Delta x^+ = 22$
- Diffuser: same mesh at inlet; for  $x \geq 2.9$ , 1.005% stretching.
- Diffusion angle:  $0 \leq \theta \leq 18^\circ$



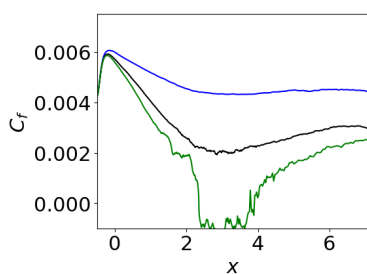


## NEXT STEP: DIFFUSER, PRESSURE GRADIENT

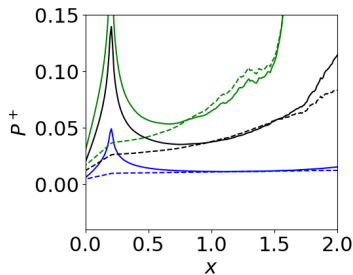
- Inlet: precursor wall-resolved LES of flow in a half-channel at  $Re_\tau = 2\,000$  ( $Re_b = 50\,000$ ).  $600 \times 150 \times 300$ ,  $0.3 < \Delta y^+ < 22$ ,  $\Delta z^+ = 11$ ,  $\Delta x^+ = 22$
- Diffuser: same mesh at inlet; for  $x \geq 2.9$ , 1.005% stretching.
- Diffusion angle:  $0 \leq \theta \leq 18^\circ$
- Upper wall: slip



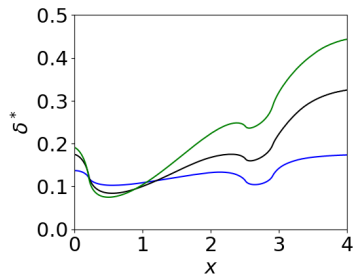
# DIFFUSER, RESULTS: $6 \leq \theta \leq 18^\circ$



(A) Skin friction.



(B) Pressure.  $P^+ = \frac{\nu \partial \bar{p} / \partial x}{u_\tau^3}$ ;  
dashed lines: Bernoulli




(C) Displacement thickness.

FIGURE: — :  $\alpha = 6^\circ$ ; — :  $\alpha = 10^\circ$ ; — :  $\alpha = 14^\circ$ ; — :  $\alpha = 18^\circ$ ; .

# CONCLUSIONS

- Machine Learning (`svr`) wall functions have been developed
- Good results for channel flow placing the wall-adjacent cell at different locations
- Good results for developing boundary layer flow
- Training the `svr` with steady or instantaneous data: **same results**
- Training nearest neighbor (Python's `scipy.spatial.KDTree`) with instantaneous data: **same results**

## REFERENCES

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