

# A-LES: WORKSHOP ON ACTIVE FLOW CONTROL, [WWW.A-LES.ORG](http://WWW.A-LES.ORG)

Chalmers, Gothenburg, 13-14 Sept 2009



# CHALMERS

## ALES for Drag Reduction of Truck-Trailers

Mohammad El-Alti, Per Kjellgren and Lars Davidson

[mohammad.el-alti@chalmers.se](mailto:mohammad.el-alti@chalmers.se)

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## Acknowledgments



Ph.D. Student	Mohammad El-Alti
Supervisor:	Lars Davidson
Co-supervisor:	Per Kjellgren
VOLVO 3P:	Linus Hjelm
SKAB:	Bengt Karlsson

# Introduction



**Aerodynamic  
drag**



**Rolling resistance**

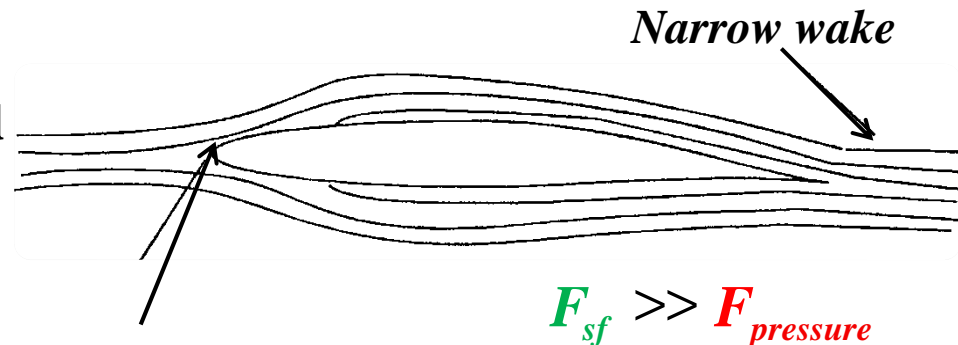
**Aerodynamic drag > Rolling resistance**

## Introduction

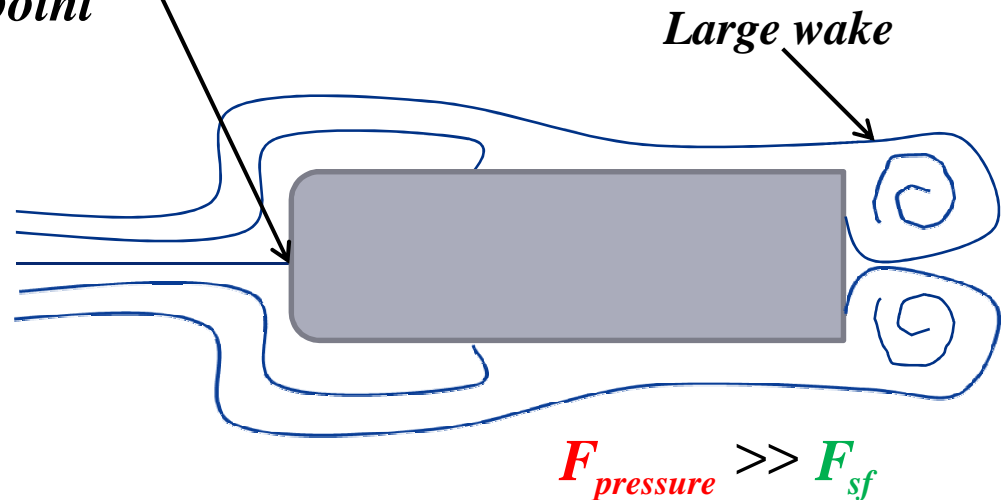
Aerodynamic drag =  
**Pressure** +  
**Skin friction**

$$F_D = \frac{1}{2} \rho A U_\infty^2 C_D$$

Streamlined  
body:

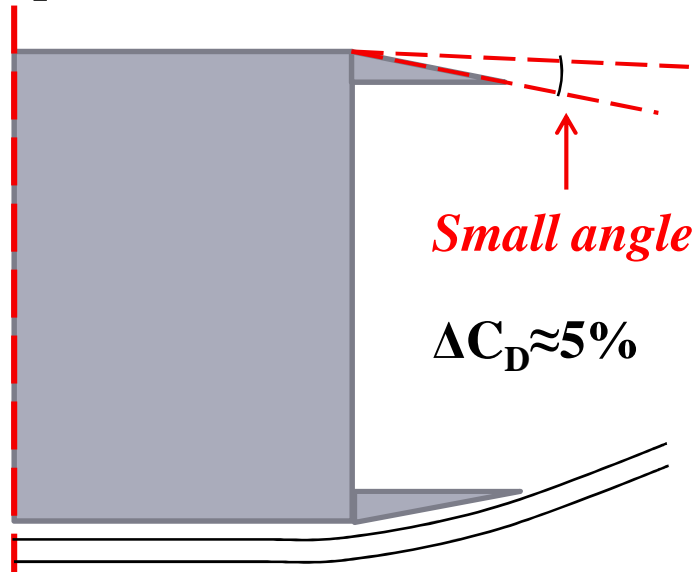


Bluff body:

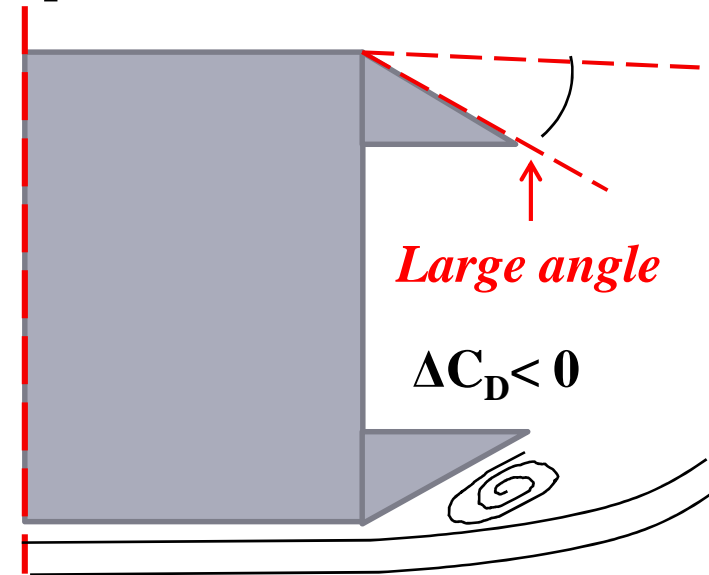


## Idea: Passive Flow Control

*Step 1:*



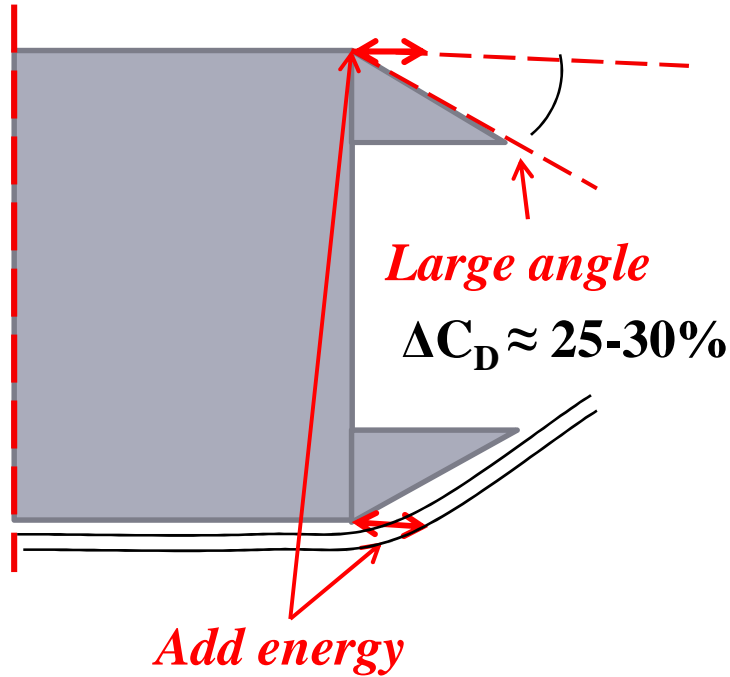
*Step 2:*



**Rear end of a Bluff body**

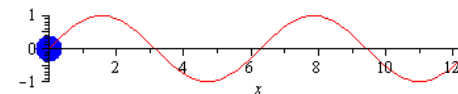
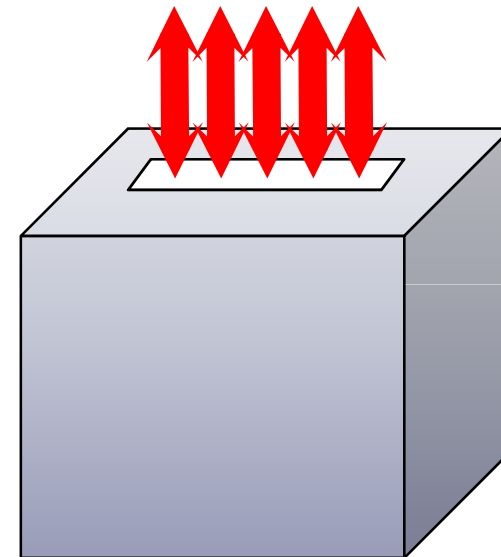
## Idea: Passive + Active Flow Control

Step 3:



$\Delta C_D \approx 25-30\% \longleftrightarrow 15\% \text{ Fuel save}$

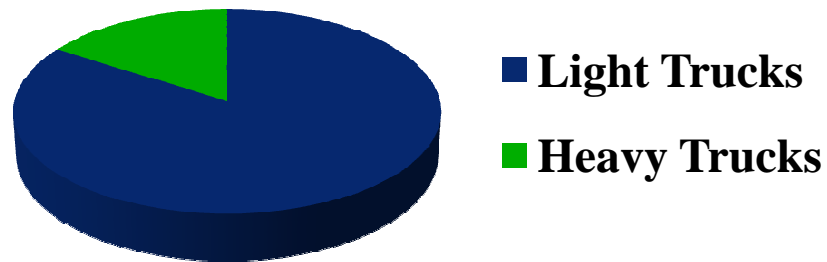
Actuator



$$u(t) = A \sin(2\pi Ft)$$

## Background

How much can we save in Sweden?



### Average Heavy truck

# heavy trucks	80 000
Fuel consumption :	28 liters /100 km
Annual driving distance:	50 000 km

Source: SCB

### Fuel save

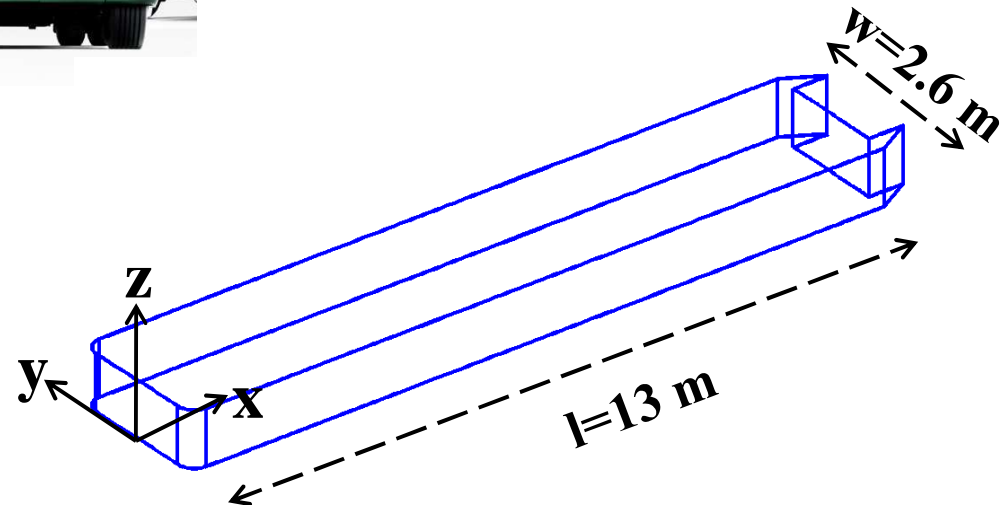
1%	➡	12 million €
15%	➡	180 million €



## First step: Truck-trailer model



Vertical spanwise slice,  
 $dz=0.2, 0.4, 0.8$  and  $1.0$  m

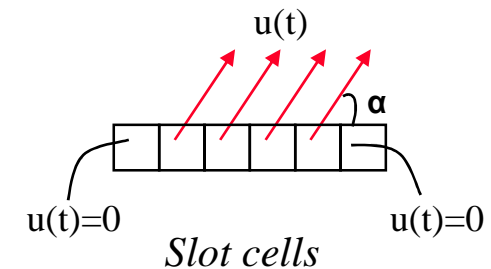


## Numerical method (1/5)

- Parameters to investigate/optimize
  - Flap length, FL
  - Flap angle, FA
  - Slot location / width
  - Slot angle, SA
  - Slot strength,  $C_\mu$
  - Slot frequency,  $F^+$
- Two codes investigated for LES
  - STAR-CD v4
    - 1 sim. sec ~ 1 week
  - FlowPhys v2
    - 1 sim. sec ~ 1 day
- *Our choice* →
- Re = 200 000, modified viscosity.

## Numerical method (2/5)

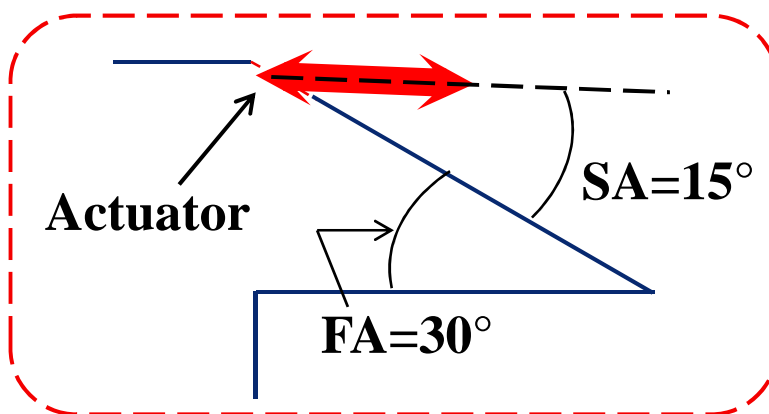
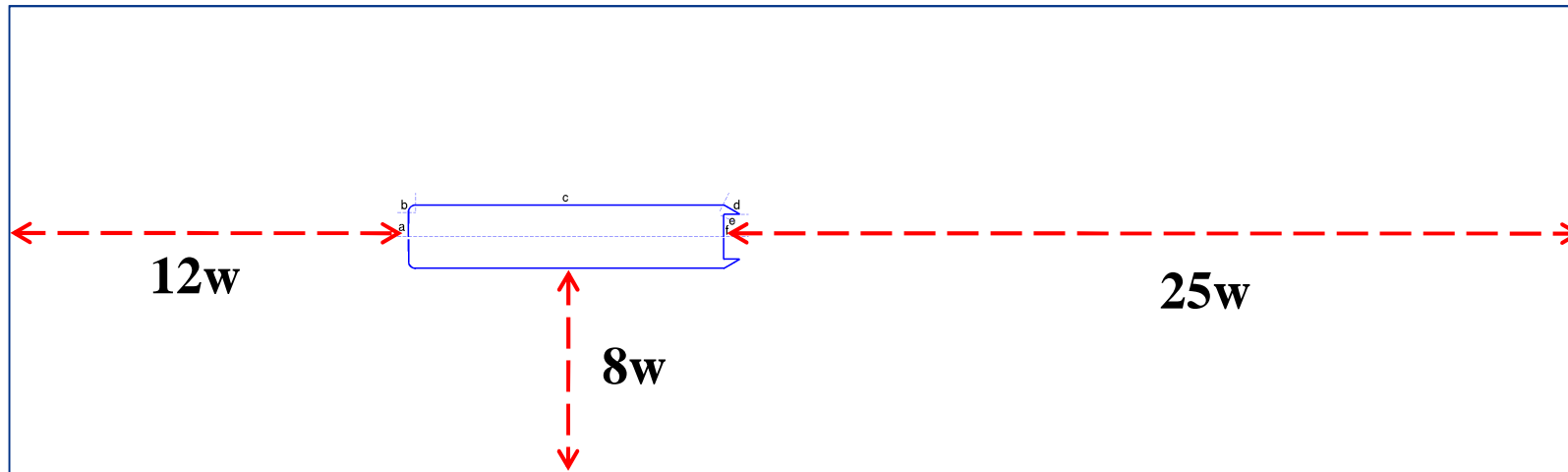
- Actuation modeling
  - Transient velocity-inlet
  - Constant spatial profile



$$\left. \begin{aligned} J_{rms} &= \int \rho u_{rms}^2 dh = \rho u_{rms}^2 \Delta h \\ C_{\mu rms} &= \frac{u_{rms}^2 \Delta h}{chord \frac{1}{2} u_{\infty}^2} \\ F^+ &= \frac{F \cdot X_{TE}}{U_{\infty}} \end{aligned} \right\}$$

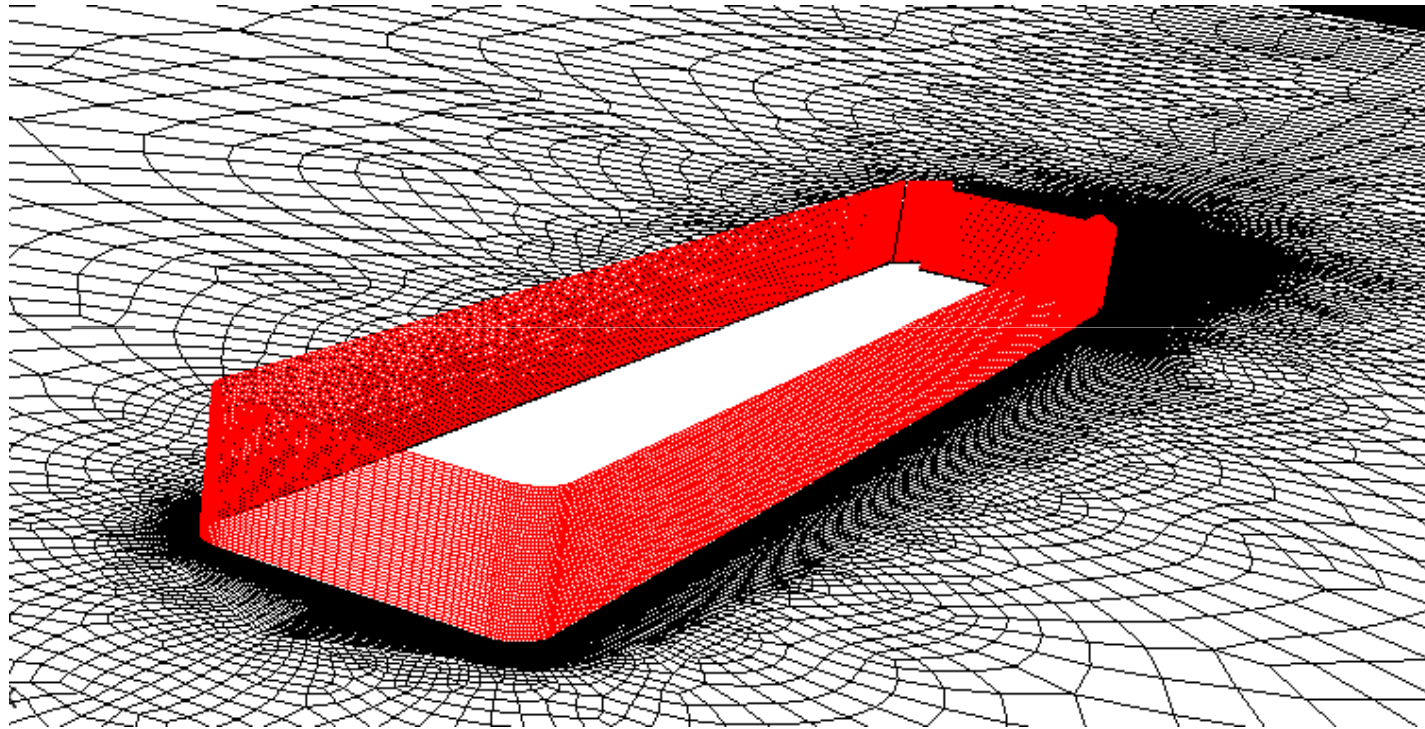
$$u(t) = \sqrt{2} u_{rms} \sin(2\pi F t)$$

## Numerical method (3/5)

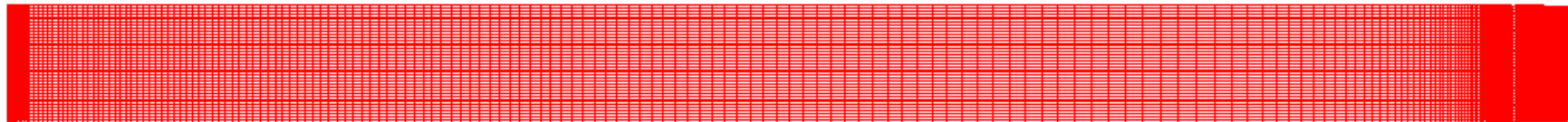


AFC Parameter	Value
$FL$	$0.75\text{ m}$
$FA$	$30\text{ deg}$
$SA$	$15\text{ deg}$
$C_\mu$	$1.0\%$
$F^+$	$0.3$

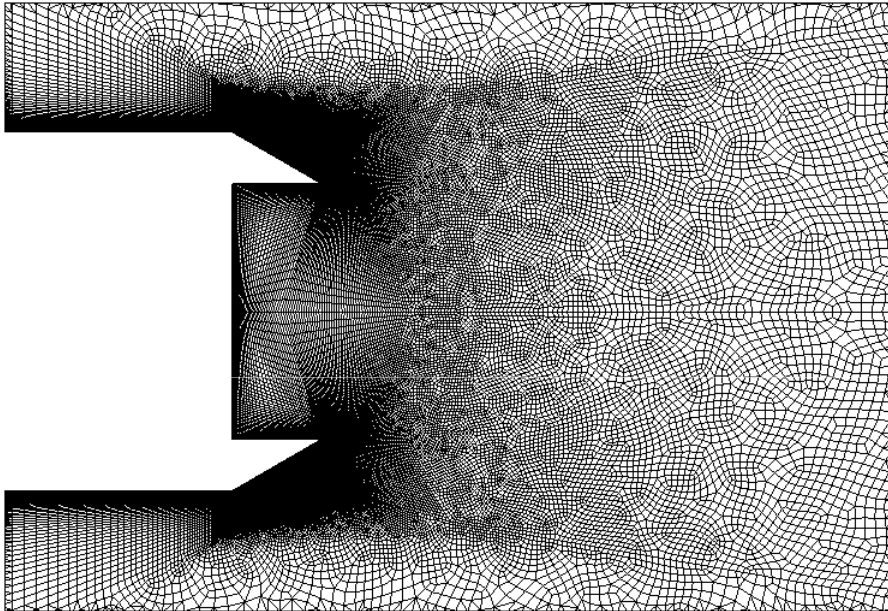
## Numerical method (4/5)



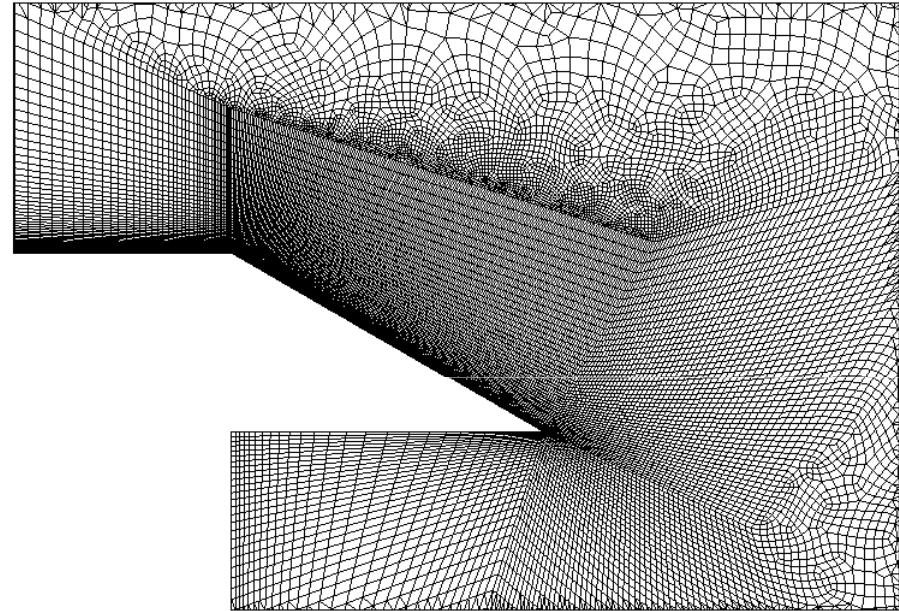
**#cells:  
1.5-3.3  
million**



## Numerical method (5/5)



**Wake region**



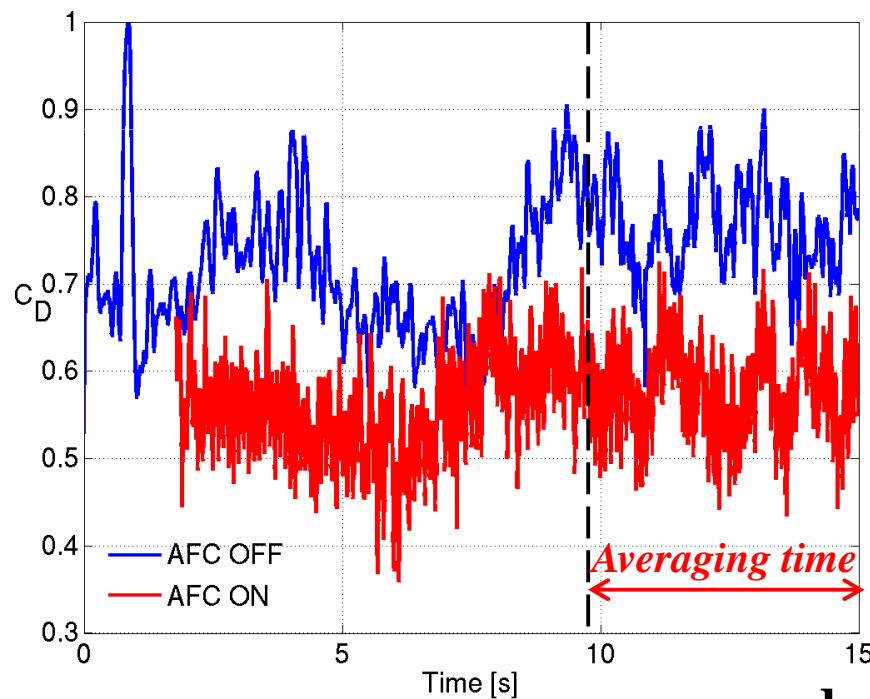
**Flap region**

	$y^+$	$\Delta x^+$	$\Delta z^+ LR$	$\Delta z^+ HR$
<i>Max</i>	2	30	150	75
<i>Mean</i>	1	20	100	50

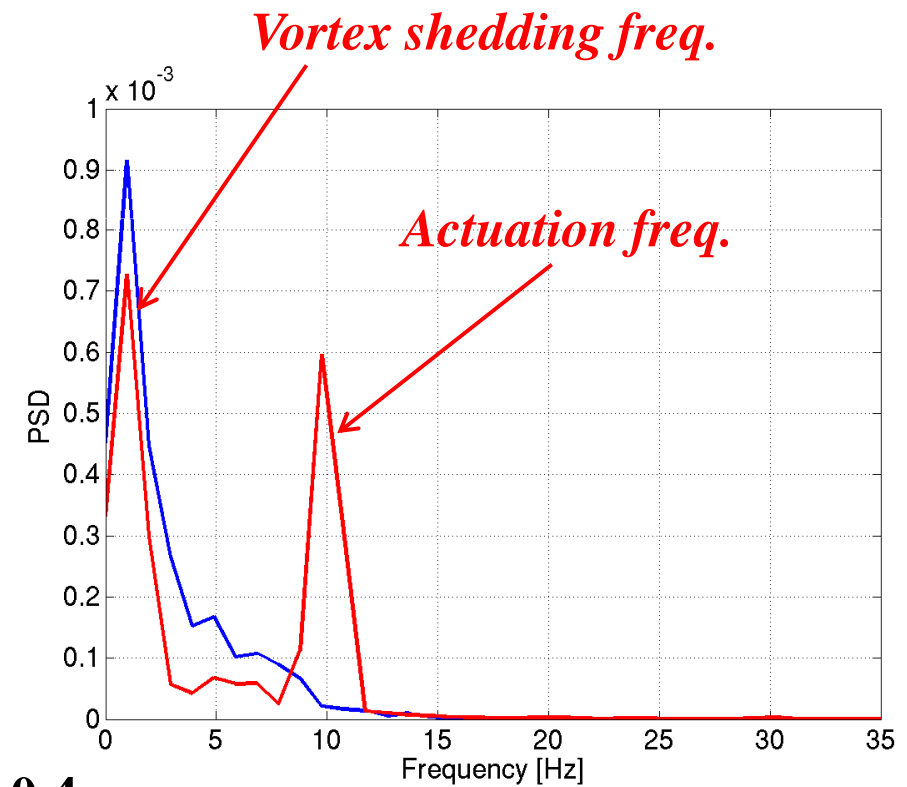
# Results

## Drag history and frequency spectra

$$C_D=0.76 \quad C_D=0.57 \quad \Delta C_D=25\%$$



$dz=0.4m$

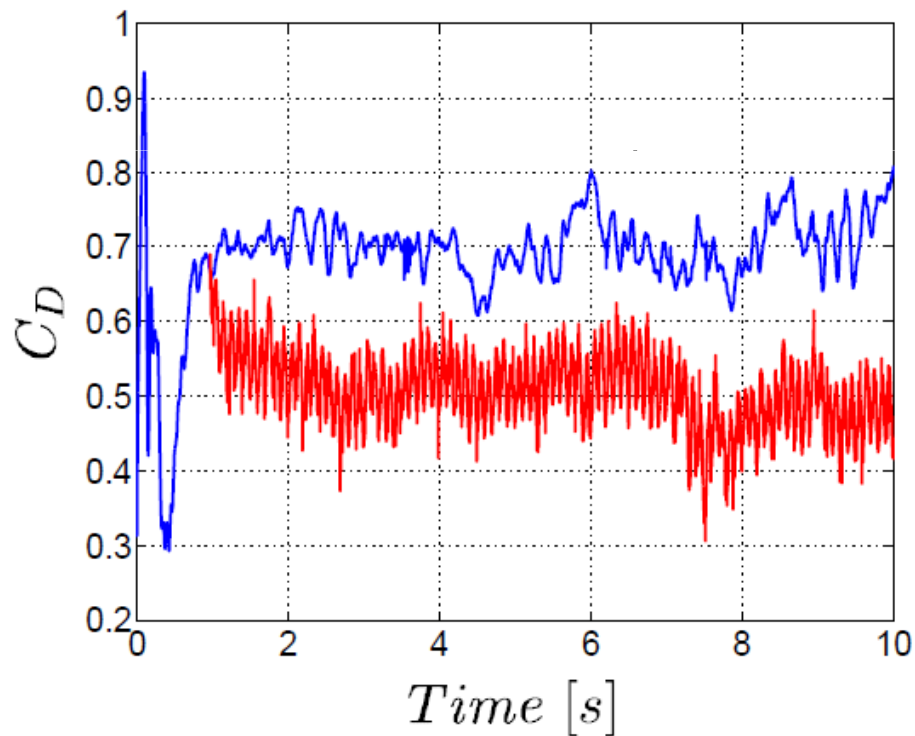




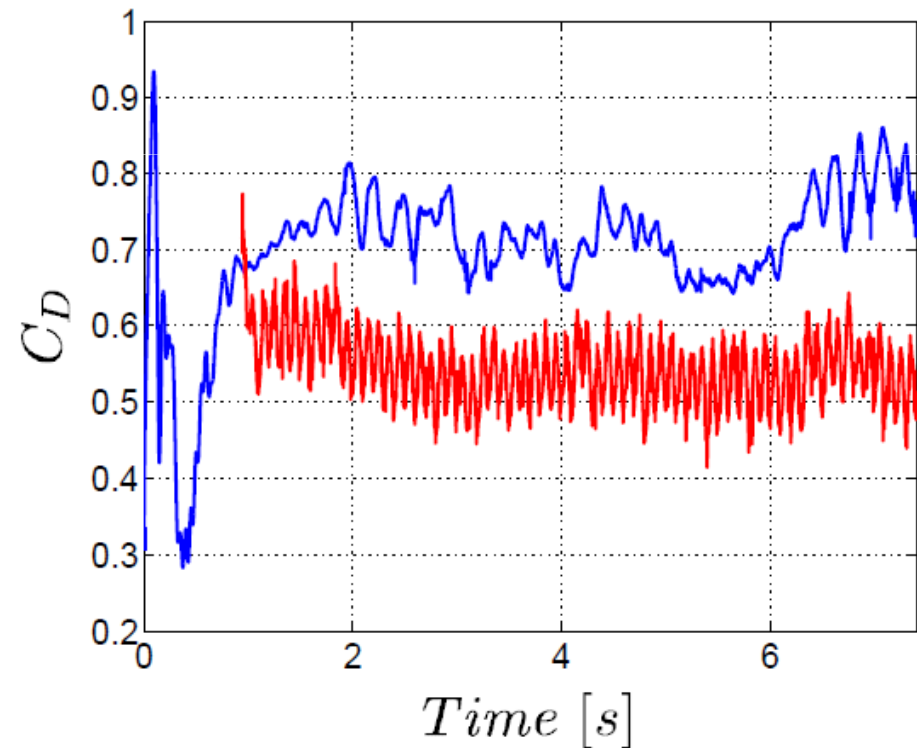
## Results

### Drag history of different spanwise domain sizes

$dz=0.8\text{m}$ ,  $\Delta C_D=30\%$



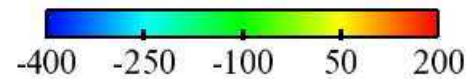
$dz=1.0\text{m}$ ,  $\Delta C_D=25\%$



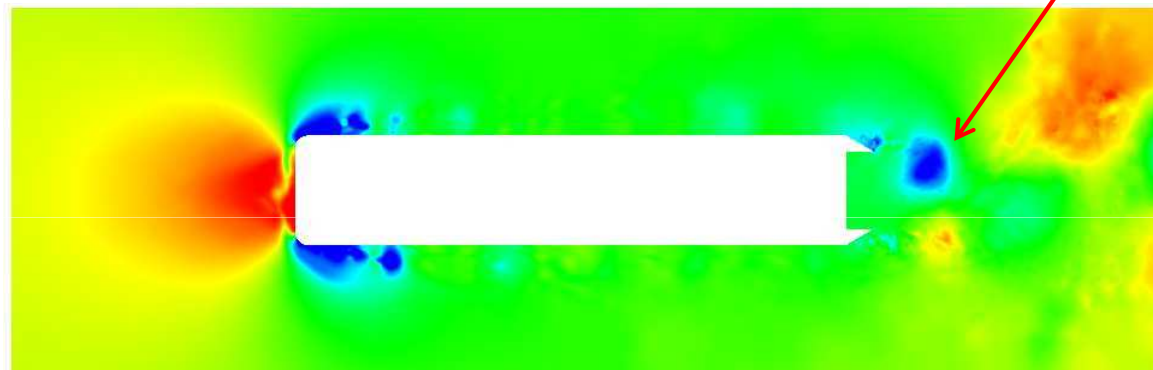


## Results

Instantaneous pressure



AFC OFF

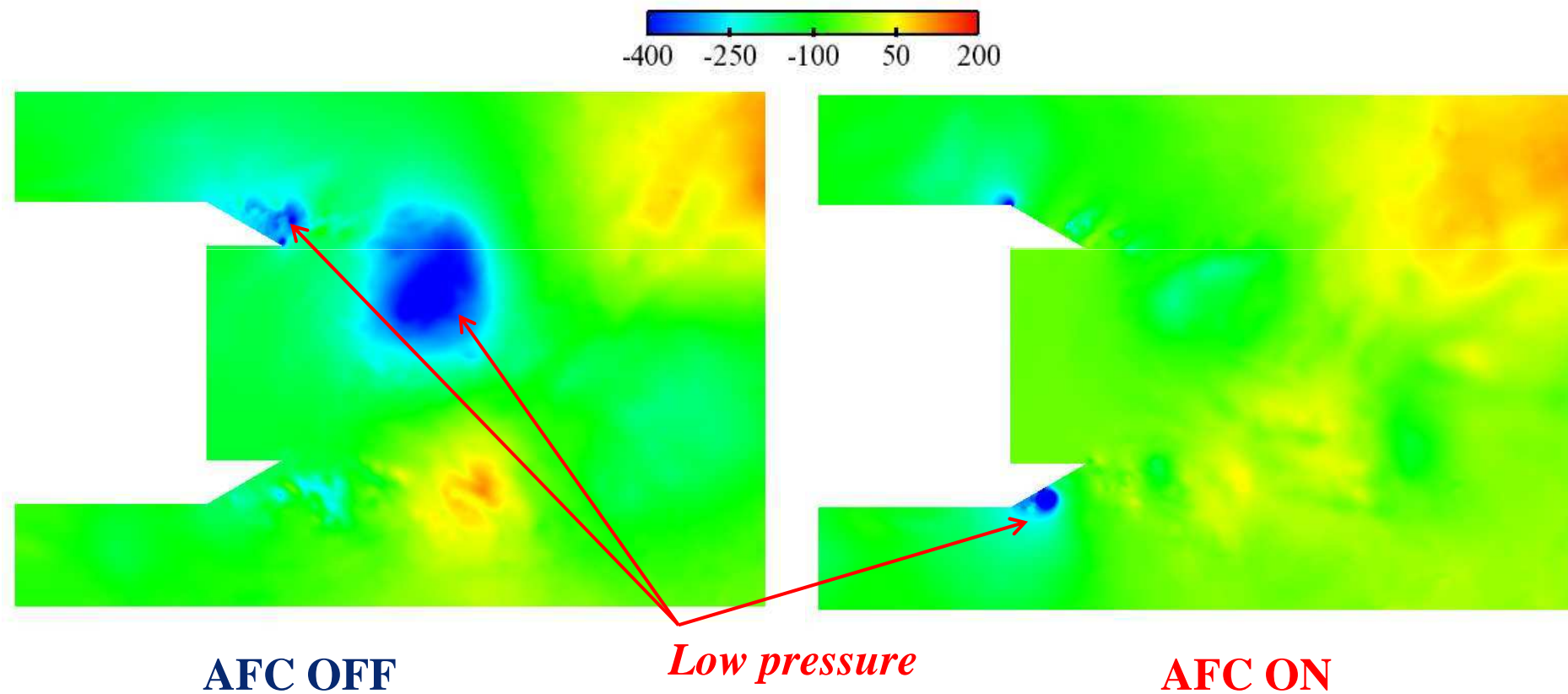


AFC ON



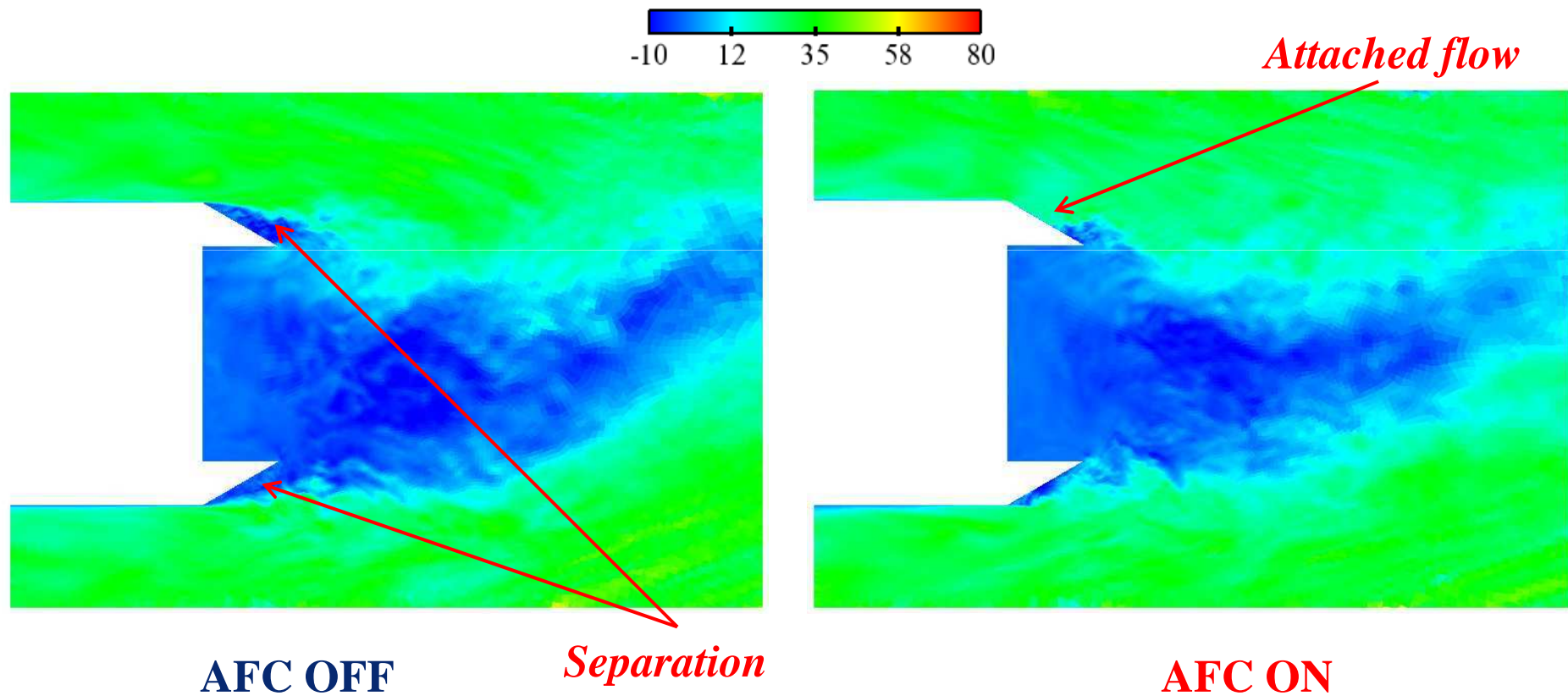
## Results

Instantaneous pressure, zoom around wake region



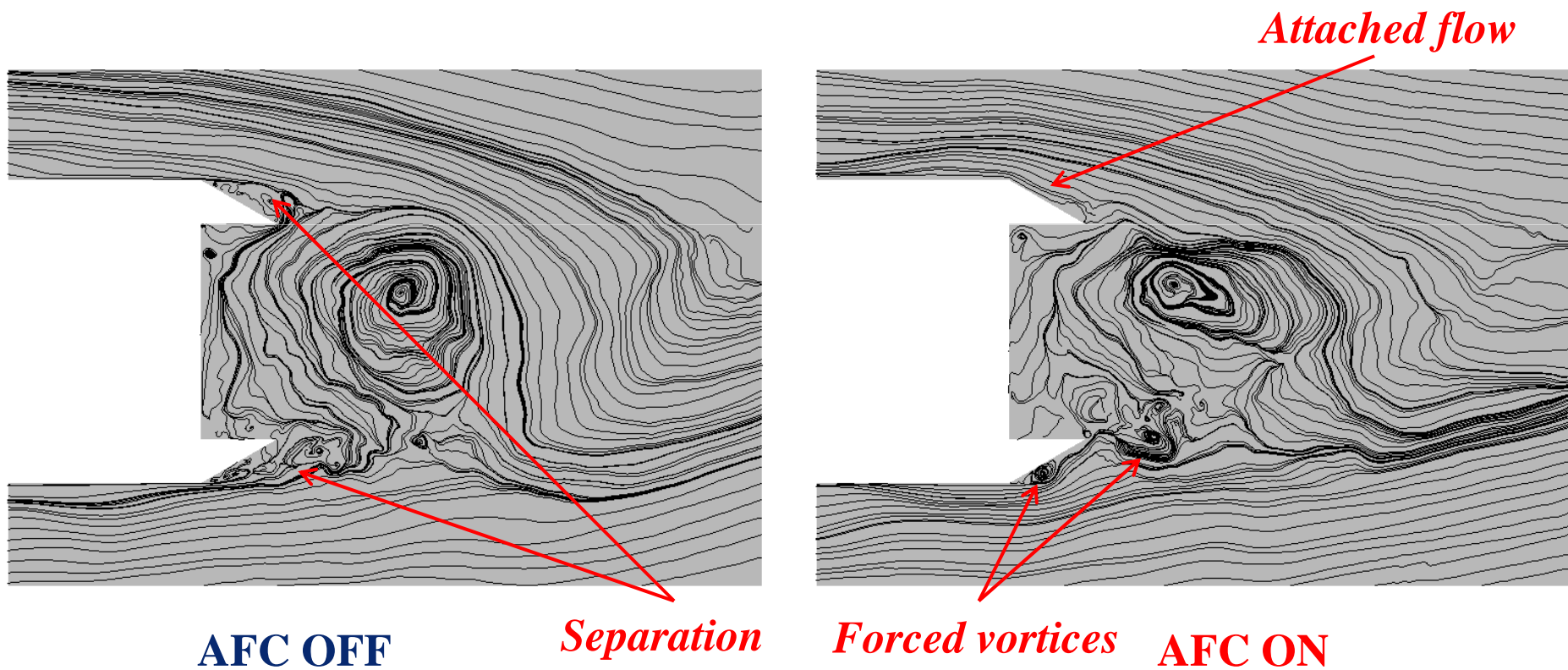
## Results

Instantaneous u-velocity, zoom around wake region



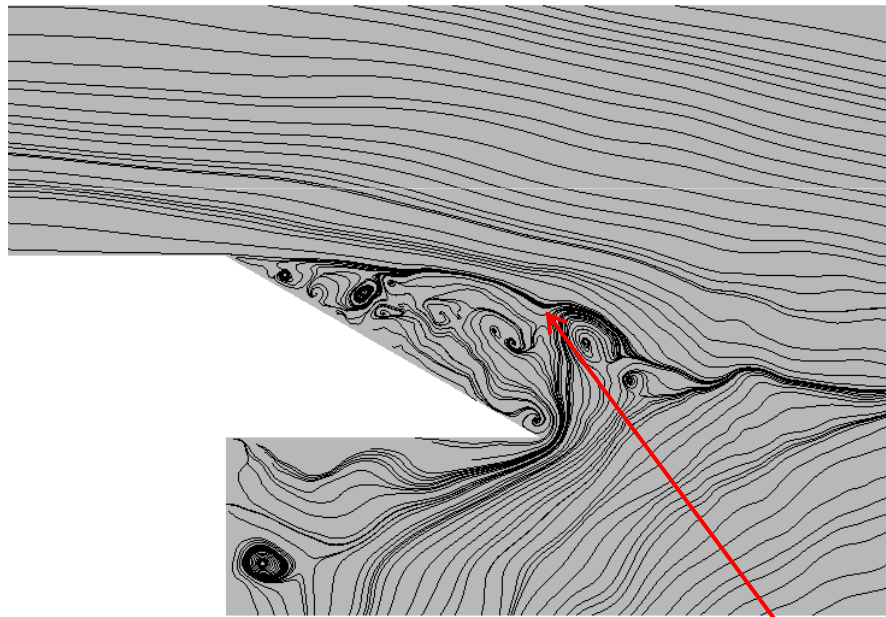
## Results

Instantaneous streamlines, zoom around wake region



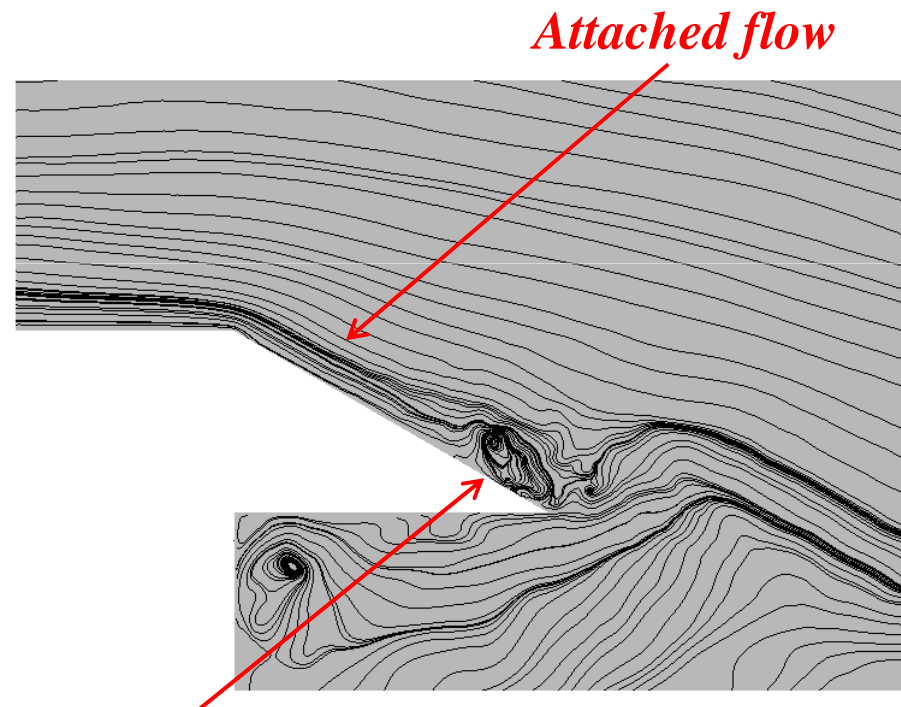
## Results

Instantaneous streamlines, zoom around flap region



**AFC OFF**

*Separation*



*Forced vortices*

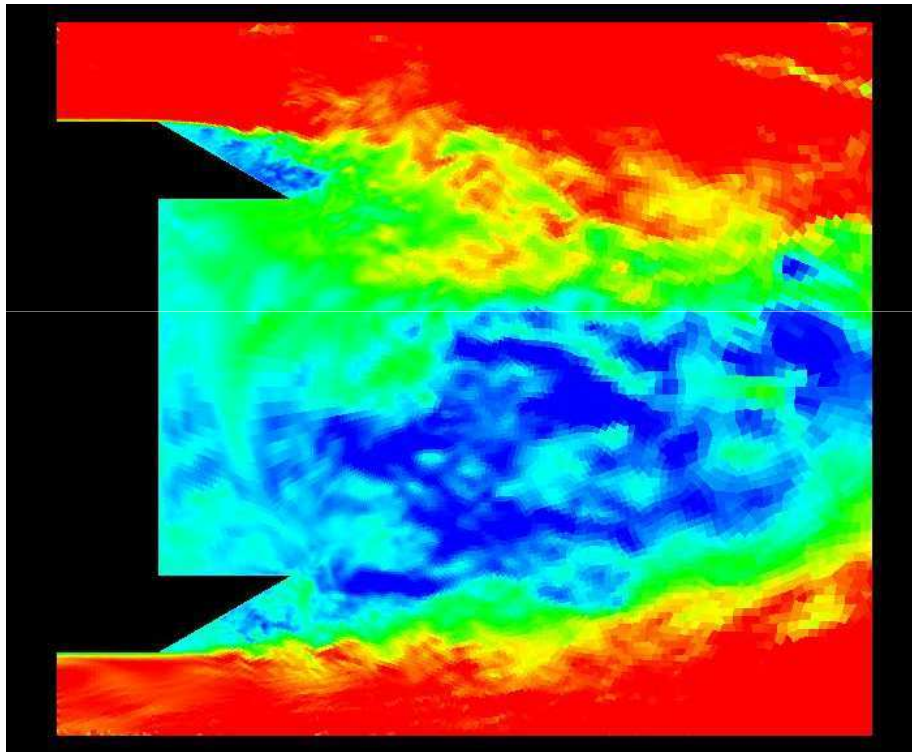
**AFC ON**

*Attached flow*

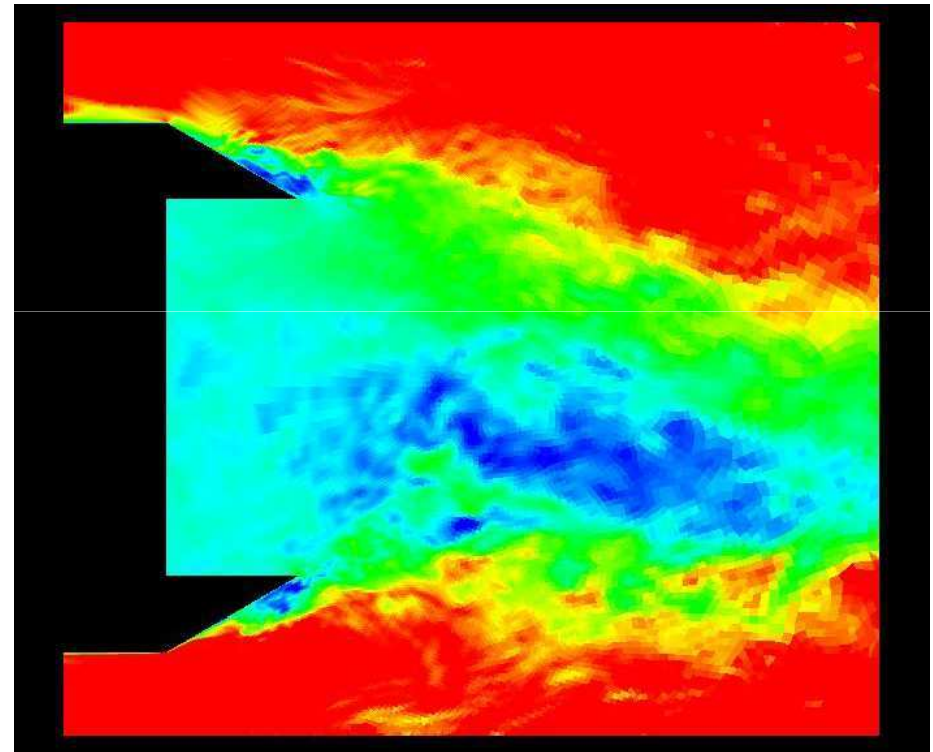


## Results

Movie: u-velocity, zoom around wake region



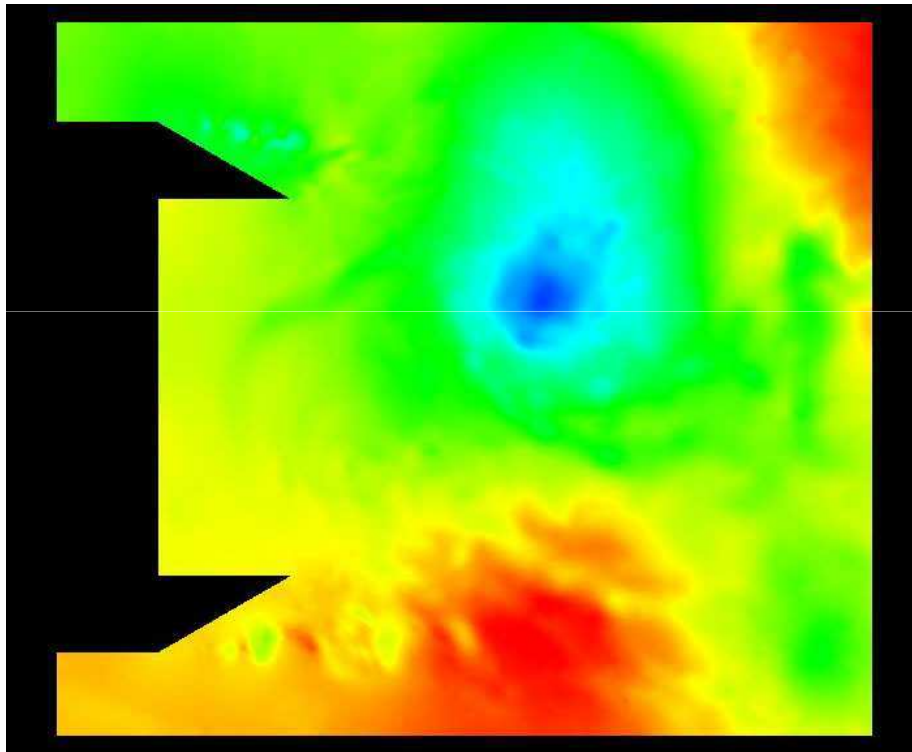
**AFC OFF**



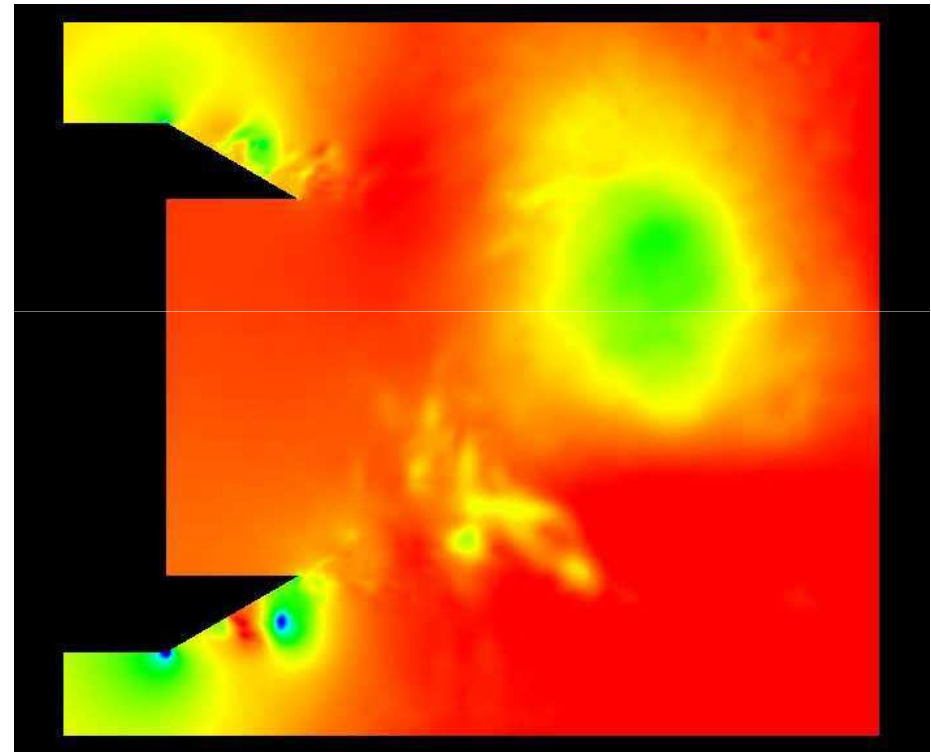
**AFC ON**

## Results

**Movie: pressure, zoom around wake region**



**AFC OFF**

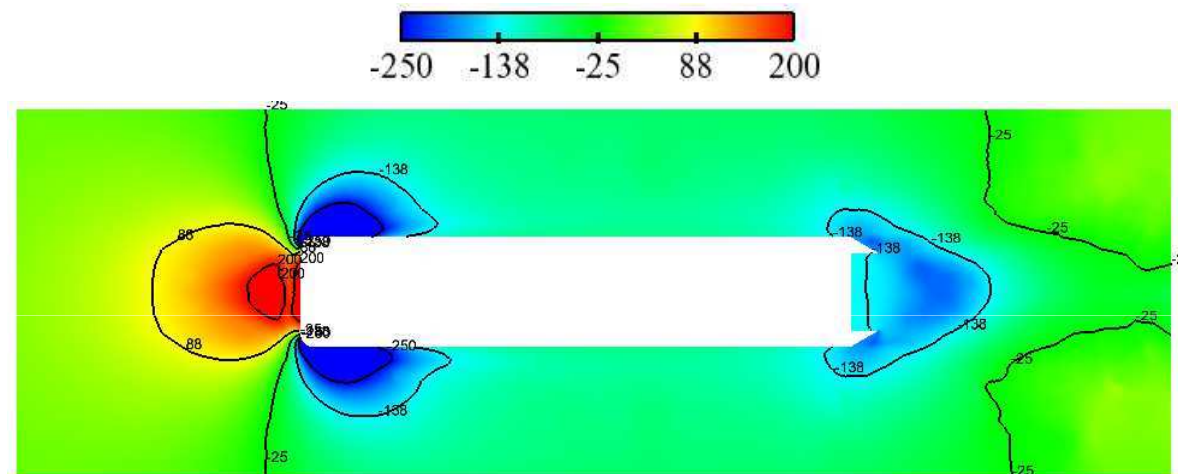


**AFC ON**

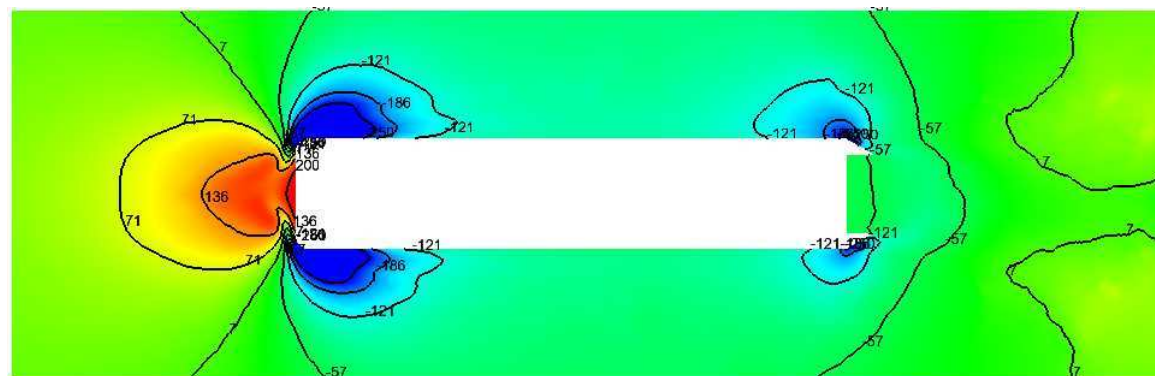
## Results

Time-averaged pressure, zoom around wake region

AFC OFF



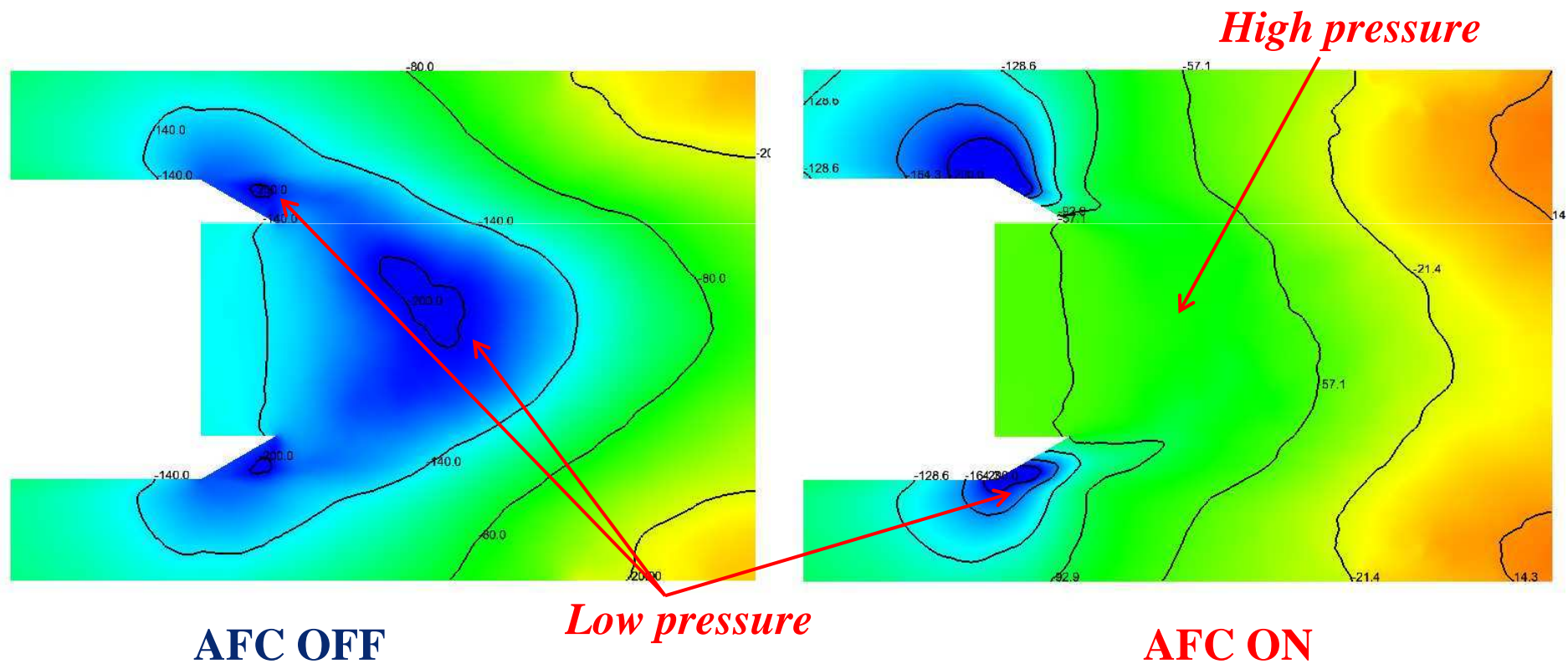
AFC ON





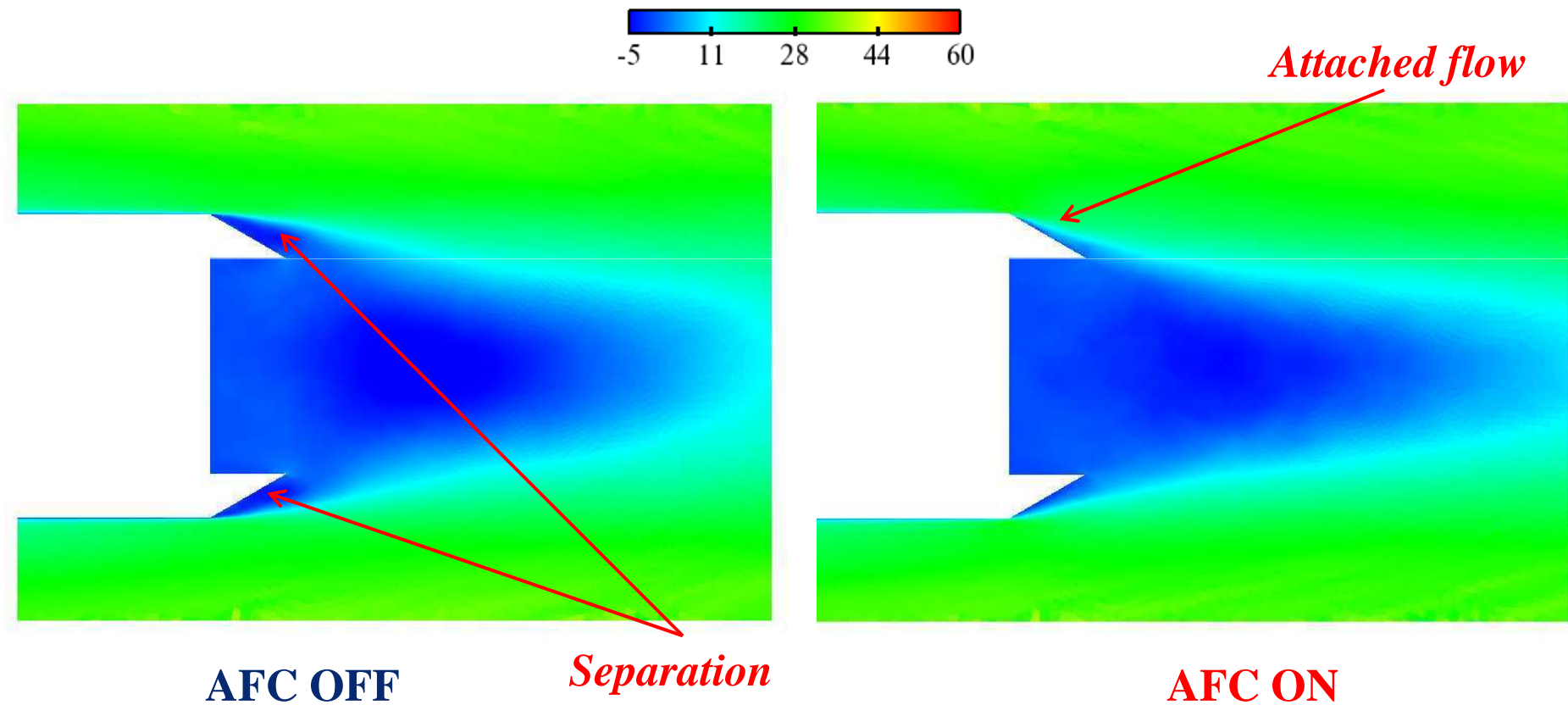
## Results

Time-averaged pressure, zoom around wake region



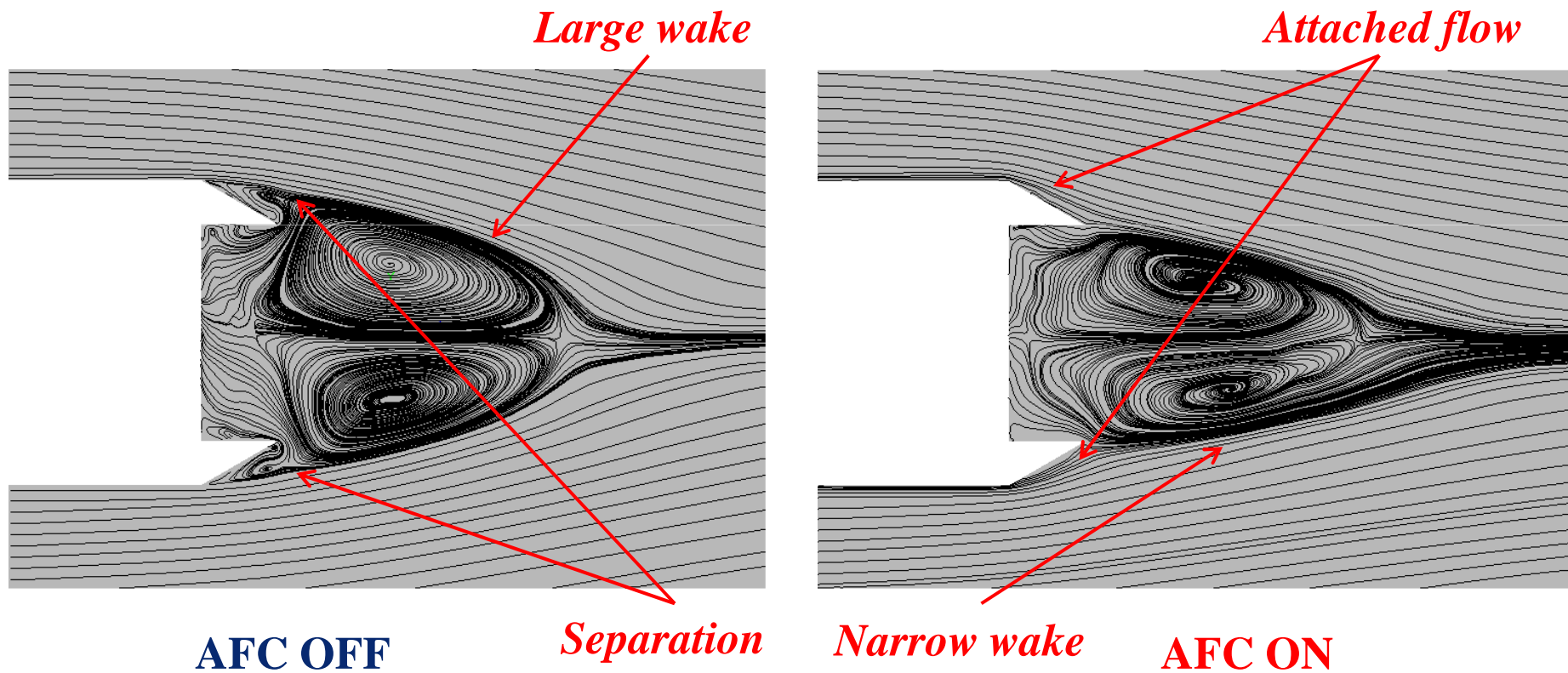
## Results

Time-averaged u-velocity, zoom around wake region



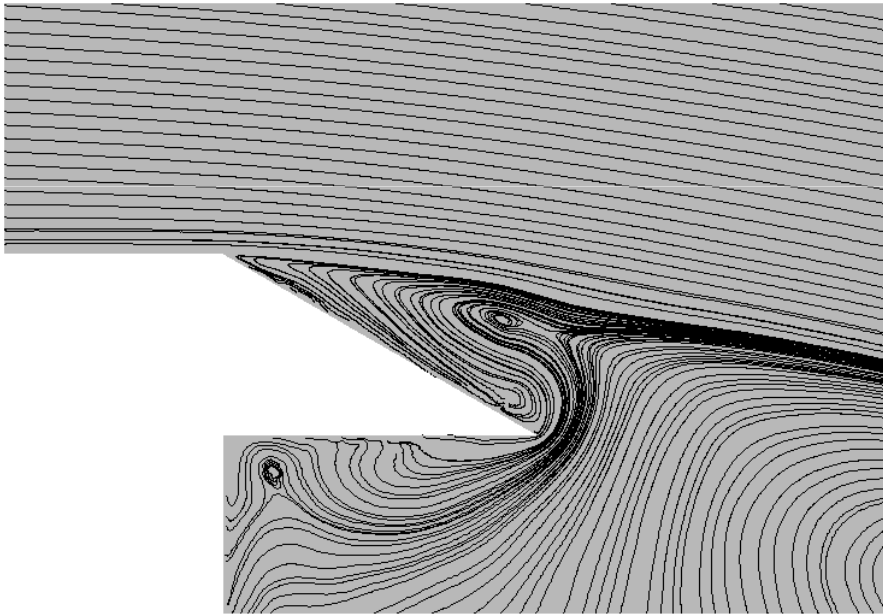
## Results

Time-averaged streamlines, zoom around wake region

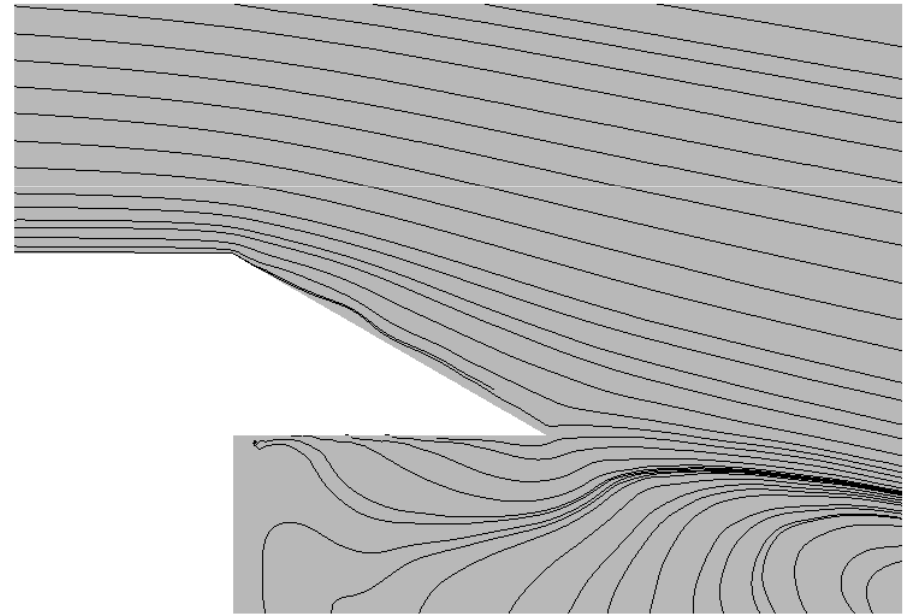


## Results

**Time-averaged streamlines, zoom around flap region**



**AFC OFF**

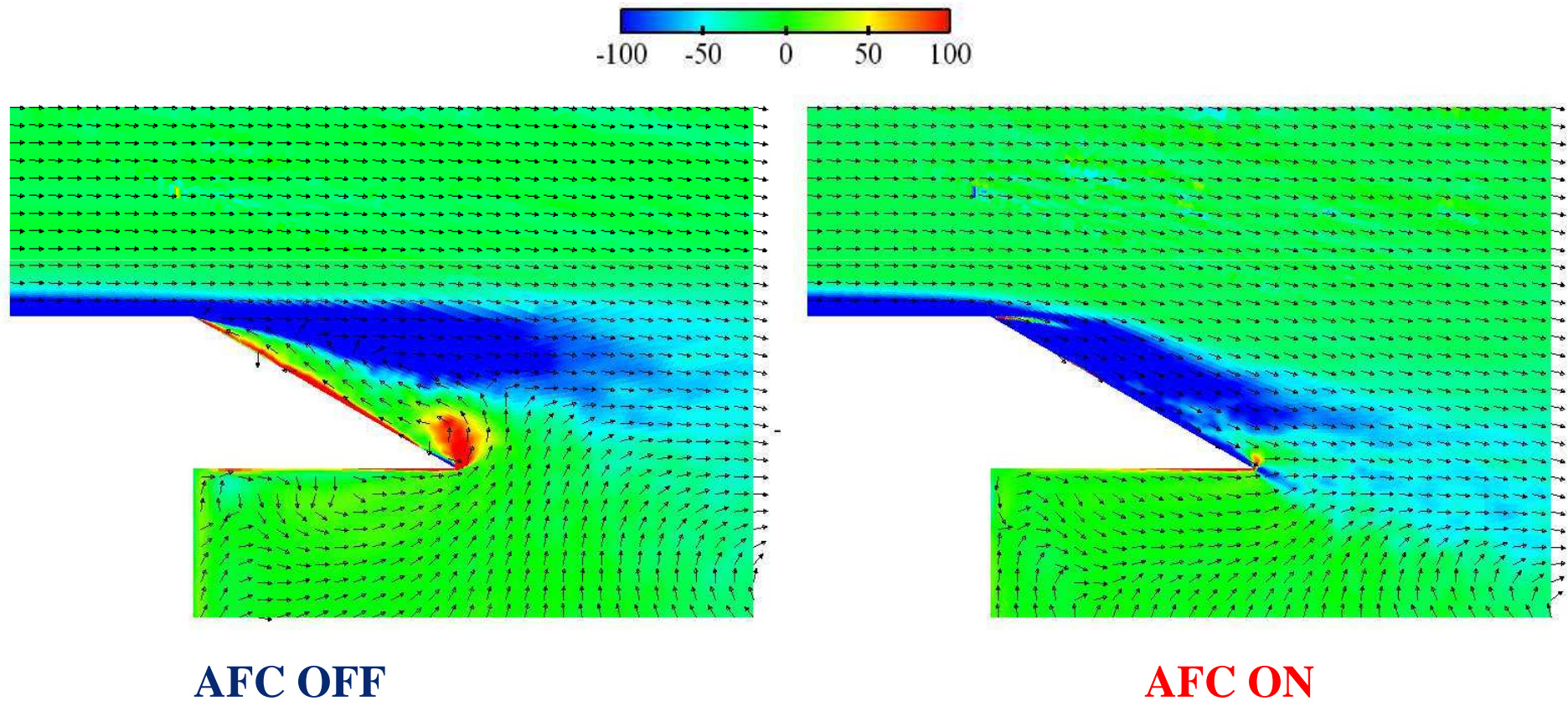


**AFC ON**



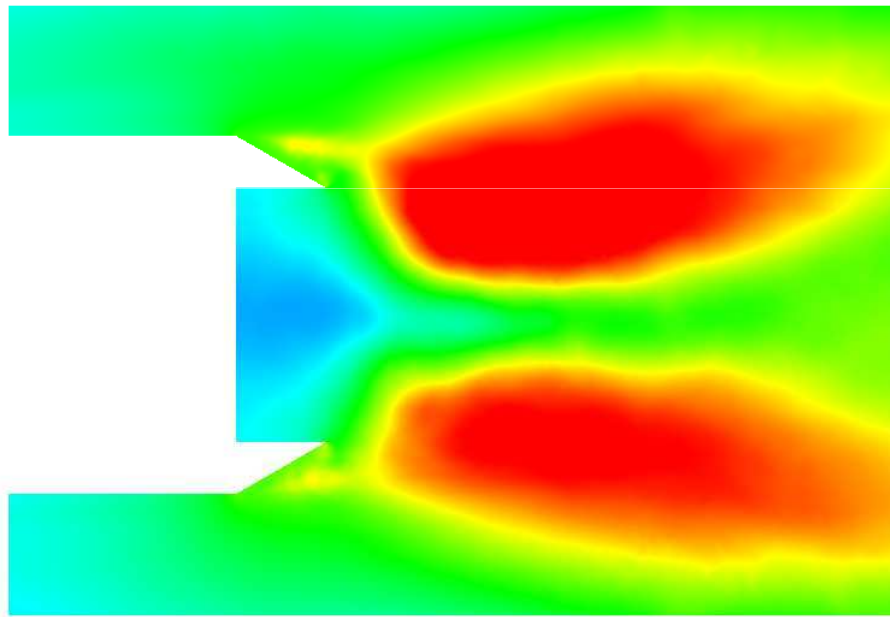
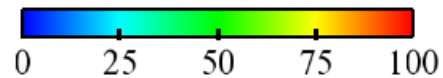
## Results

Time-averaged vorticity, zoom around flap region

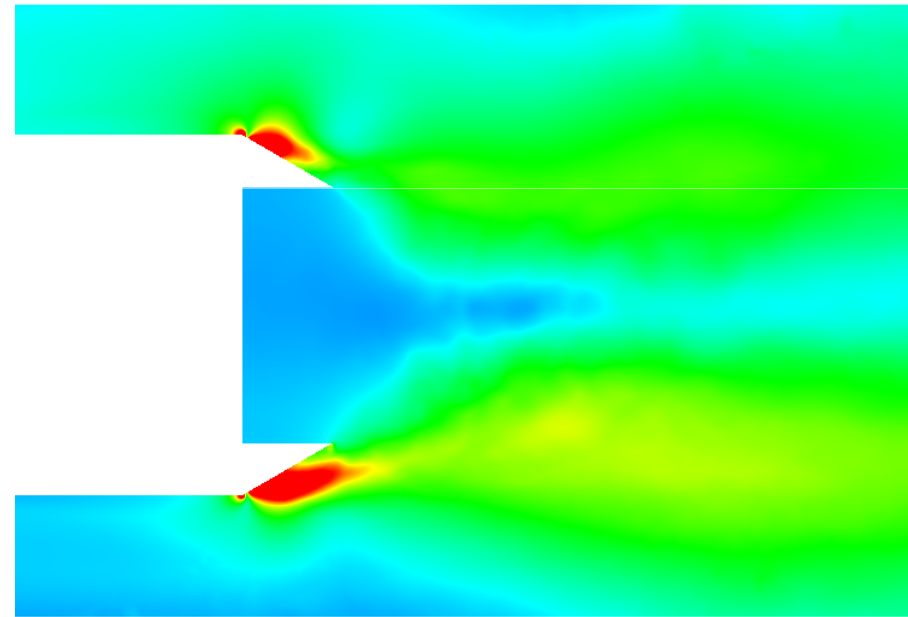


## Results

**RMS of pressure, zoom around wake region**



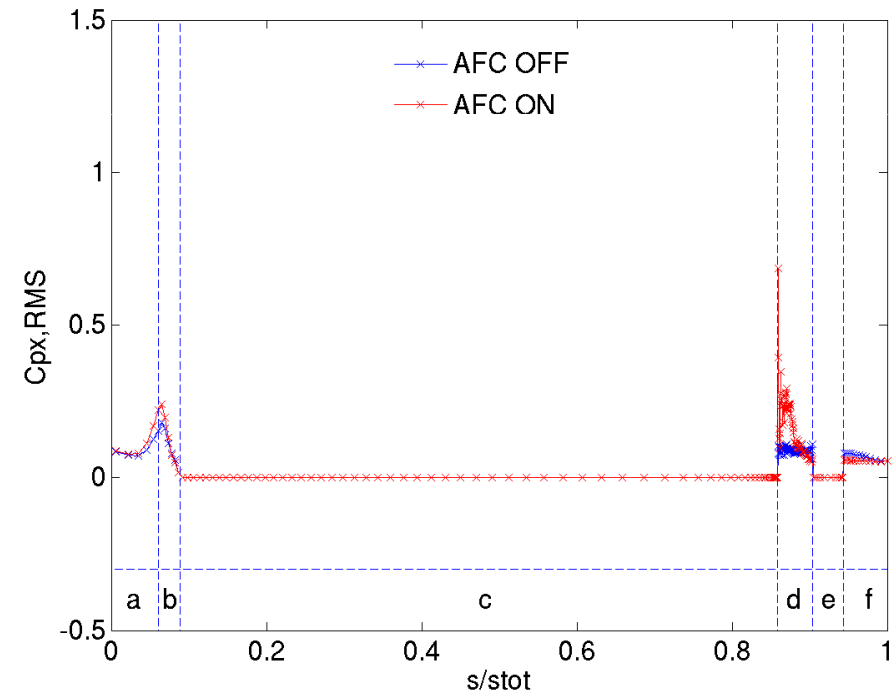
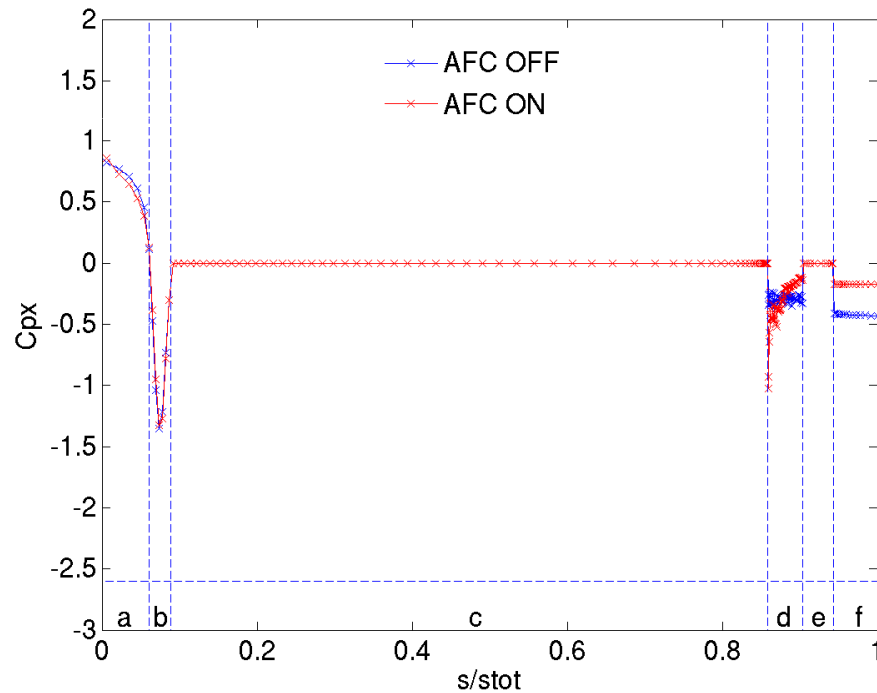
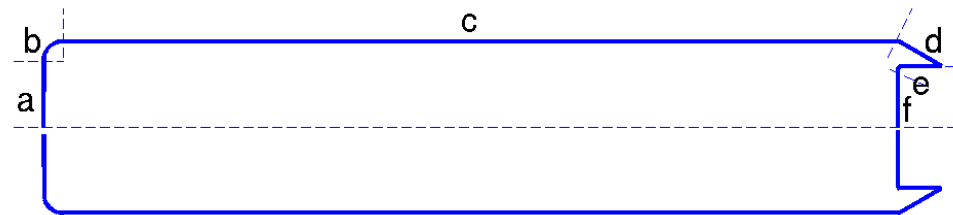
**AFC OFF**



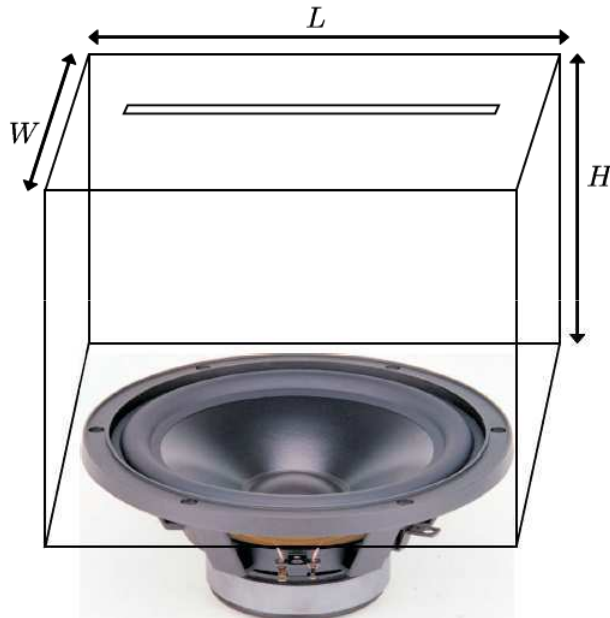
**AFC ON**

# Results

## Surface pressure distribution



## Second step: Built and test of actuator



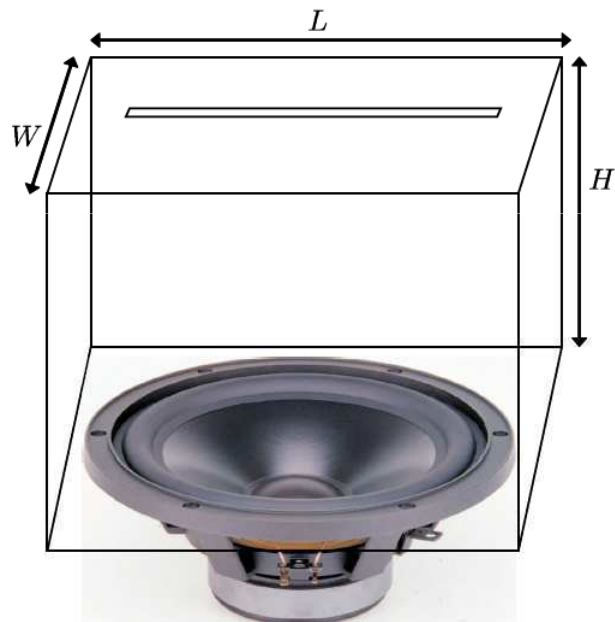
**Simple synthetic-jet actuator**



**Max velocity – max  $C_\mu$ ?  
Slotwidth, speaker, cavity volume?**



## Third step: Build the prototype model



**Simple synthetic-jet actuator**



## Forth step: Full scale test of the prototype model

**VOLVO**



**November  
2009**

**Hällered proving ground**

## Conclusions

- Drag reduction of 25-30%
  - Flow reattachment
  - Narrower wake size
  - Less intensive wake
  - Increased base pressure
- AFC works well for bluff bodies

## Future work

- Optimization using RSM
- Parallellizing FlowPhys and/or using DES with STAR-CD
  - Fully 3D bluff body with AFC
  - Fully detailed truck-trailer with AFC
- More analysis on the drag reduction mechanism
- Study more applications of AFC

**Thank you!**  
**Questions**

